

# Numerical Heat Transfer

## (数值传热学)

### Chapter 3 Numerical Methods for Solving Diffusion Equation and their Applications (2)



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## 3.4 TDMA & ADI Methods for Solving ABEs

### 3.4.1 TDMA algorithm (算法) for 1-D conduction problem

1. General form of algebraic equations of 1-D conduction problems

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# 3.4 TDMA & ADI Methods for Solving ABEqs

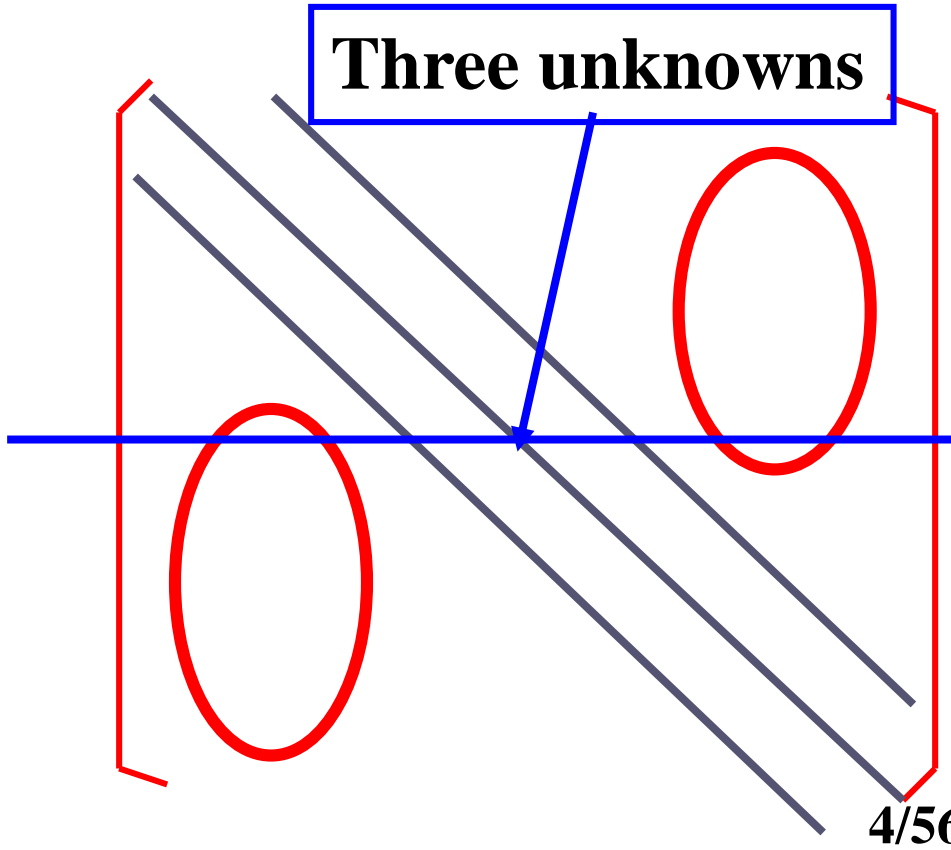
## 3.4.1 TDMA algorithm for 1-D conduction problem

### 1. General form of algebraic equations. of 1-D conduction problems

The ABEqs for steady and unsteady ( $f > 0$ ) problems take the form

$$a_P T_P = a_E T_E + a_W T_W + b$$

The matrix (矩阵) of the coefficients is a **tri-diagonal** (三对角) one .



## 2. Thomas algorithm(算法)

The numbering method of W-P-E is humanized (人性化), but it can not be accepted by a computer!

Rewrite above equation:

$$A_i T_i = B_i T_{i+1} + C_i T_{i-1} + D_i, \quad i = 1, 2, \dots, M-1 \quad (\mathbf{a})$$

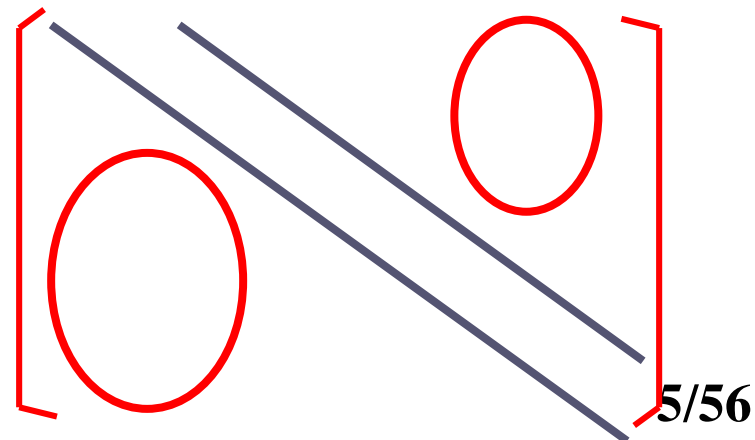
End conditions:  $i=1, C_i=0; i=M-1, B_i=0$

**(1) Elimination (消元)** – Reducing the unknowns at each line from 3 to 2

Assuming the eq. after elimination as

$$T_{i-1} = P_{i-1} T_i + Q_{i-1} \quad (\mathbf{b})$$

Coefficient has been treated to 1.



The purpose of the elimination procedure is to find the relationship between  $P_i, Q_i$  with  $A_i, B_i, C_i, D_i$ :

Multiplying Eq.(b) by  $C_i$ , and adding to Eq.(a):

$$A_i T_i = B_i T_{i+1} + \cancel{C_i T_{i-1}} + D_i \quad (\text{a})$$

$$\cancel{C_i T_{i-1}} = C_i P_{i-1} T_i + C_i Q_{i-1} \quad (\text{b})$$

$$A_i T_i - C_i P_{i-1} T_i = B_i T_{i+1} + D_i + C_i Q_{i-1}$$

**Yielding**

$$T_i = \left( \frac{B_i}{\underbrace{A_i - C_i P_{i-1}}} \right) T_{i+1} + \frac{D_i + C_i Q_{i-1}}{\underbrace{A_i - C_i P_{i-1}}}$$

**Comparing with**

$$T_{i-1} = P_{i-1} T_i + Q_{i-1}$$

$$P_i = \frac{B_i}{A_i - C_i P_{i-1}}; \quad Q_i = \frac{D_i + C_i Q_{i-1}}{A_i - C_i P_{i-1}};$$

The above equations are **recursive** – i.e.,

In order to get  $P_i$ ,  $Q_i$ ,  $P_1$  and  $Q_1$  must be known.

In order to get  $P_1$ ,  $Q_1$ , use Eq.(a)

$$A_i T_i = B_i T_{i+1} + C_i T_{i-1} + D_i, \quad i = 1, 2, \dots, M-1 \quad \text{(a)}$$

End condition:  $i=1, C_i=0; i=M-1, B_i=0$

Applying Eq.(a) to  $i=1$ , and comparing it with Eq.(b), the expressions of  $P_1$ ,  $Q_1$  can be obtained:

From  $i = 1, C_1 = 0, A_1 T_1 = B_1 T_2 + D_1$

$$T_1 = \frac{B_1}{A_1} T_2 + \frac{D_1}{A_1} \longrightarrow P_1 = \frac{B_1}{A_1}; Q_1 = \frac{D_1}{A_1}$$

**(2) Back substitution(回代) – Starting from M1 via Eq.(b) to get  $T_i$  sequentially (顺序地)**

$$T_{M1} = P_{M1} T_{M1+1} + Q_{M1}, \quad P_i = \frac{B_i}{A_i - C_i P_{i-1}};$$

**End condition:  
 $i = M1, B_i = 0$**

$$\longrightarrow P_{M1} = 0$$

$$T_{M1} = Q_{M1} \boxed{T_{i-1} = P_{i-1} T_i + Q_{i-1}} \text{ to get: } T_{M1-1}, \dots, T_2, T_1.$$



### 3. Implementation of Thomas algorithm for 1<sup>st</sup> kind B.C.

For 1<sup>st</sup> kind B.C., the solution region is from  $i=2, \dots$  to  $M1-1=M2$ .

Applying Eq.(b) to  $i=1$  with given  $T_{1,\text{given}}$ :

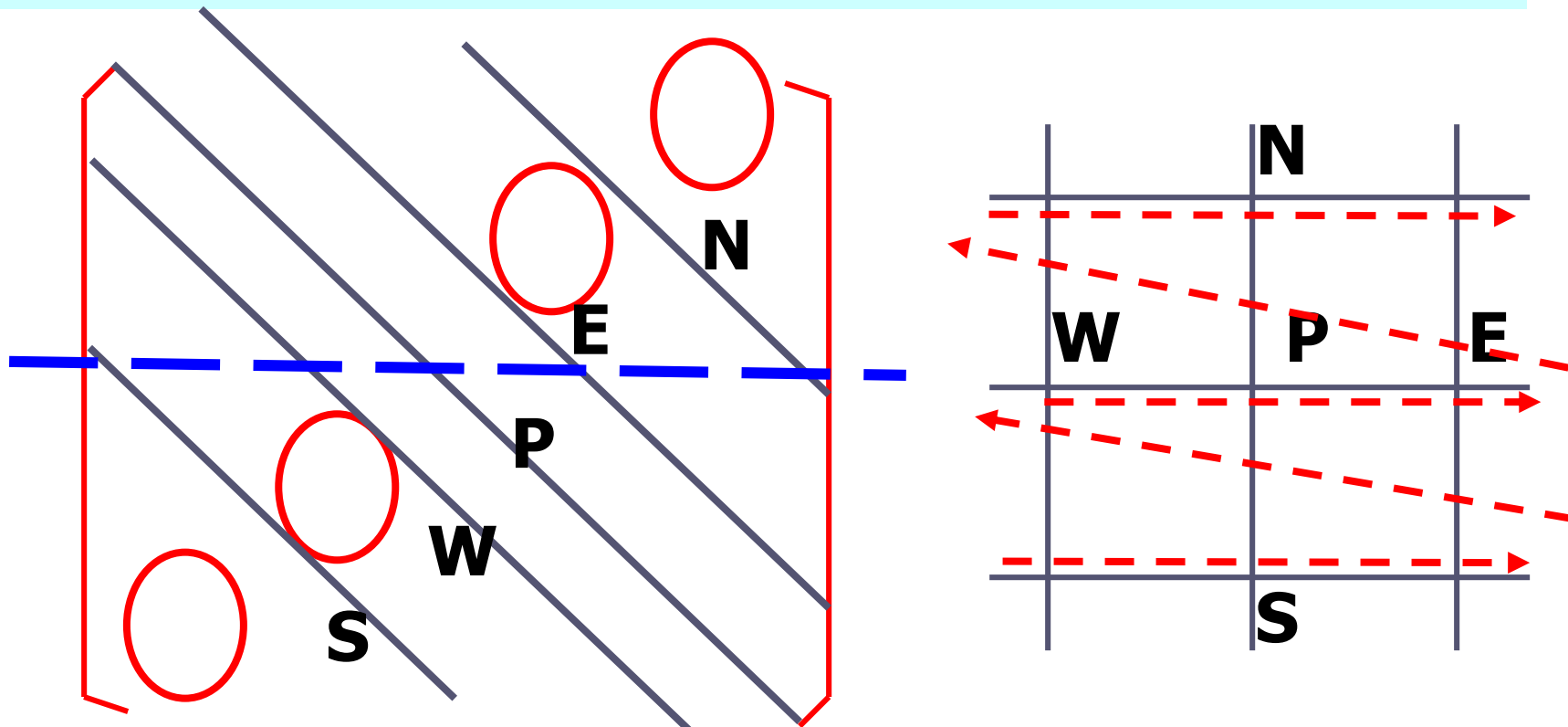
$$T_1 = P_1 T_2 + Q_1 \quad \longrightarrow \quad P_1 = 0; \quad Q_1 = T_{1,\text{given}}$$

Because  $T_{M1}$  is known, back substitution should be started from  $M_2$ :  $T_{M2} = P_{M2} T_{M1} + Q_2$

When the ASTM is adopted to deal with B.C. of 2<sup>nd</sup> and 3<sup>rd</sup> kind, **the numerical B.C. for all cases is regarded as 1<sup>st</sup> kind**, and the above treatment should be adopted.

# 3.4.2 ADI method for solving multi-dimensional problem

## 1. Introduction to the matrix of 2-D problem



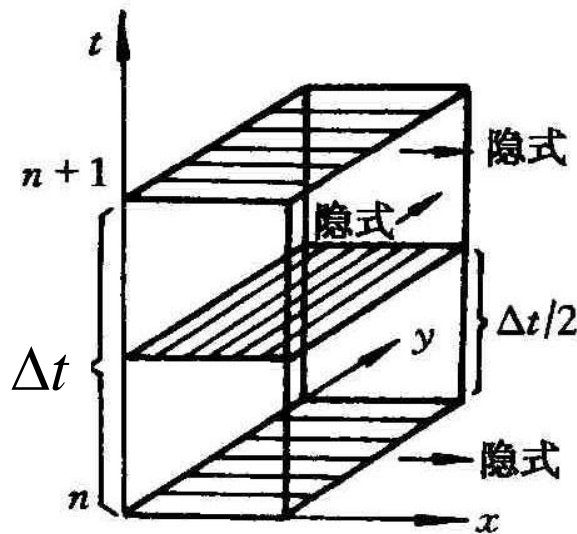
1-D storage (一维存储) of variables and its relation to matrix coefficients

# Numerical methods for solving ABEqs. of 2-D problems.

**(1) Penta-diagonal algorithm(PDMA,五对角阵算法)**

**(2) Alternative(交替的)-direction Implicit (ADI, 交替方向隱式方法)**

## 2. 3-D Peaceman-Rachford ADI method



**2-D ADI**

Dividing  $\Delta t$  into three uniform parts  
 In the 1st  $\Delta t / 3$  implicit in x direction,  
 and explicit in y, z directions;  
 In the 2<sup>nd</sup> and 3<sup>rd</sup>  $\Delta t / 3$  implicit in  
 y, z direction, respectively.

Set  $u_{i,j,k}$ ,  $v_{i,j,k}$  the temporary (临时的) solutions at two sub-time levels

$\delta_x^2 T_{i,j,k}^n$  -CD for 2<sup>nd</sup> derivative at n time level in x direction

1<sup>st</sup> sub-time level  $\frac{u_{i,j,k} - T_{i,j,k}^n}{\Delta t / 3} = a(\delta_x^2 u_{i,j,k} + \delta_y^2 T_{i,j,k}^n + \delta_z^2 T_{i,j,k}^n)$

2<sup>nd</sup> sub-time level:  $\frac{v_{i,j,k} - u_{i,j,k}^n}{\Delta t / 3} = a(\delta_x^2 u_{i,j,k} + \delta_y^2 v_{i,j,k} + \delta_z^2 u_{i,j,k}^n)$

3<sup>rd</sup> sub-time level  $\frac{T_{i,j,k}^{n+1} - v_{i,j,k}^n}{\Delta t / 3} = a(\delta_x^2 v_{i,j,k} + \delta_y^2 v_{i,j,k}^n + \delta_z^2 T_{i,j,k}^{n+1})$

**It's obvious that this solution procedure is not fully implicit, and the time step is limited by following stability condition:**

$$a\Delta t \left( \frac{1}{\Delta x^2} + \frac{1}{\Delta y^2} + \frac{1}{\Delta z^2} \right) \leq 1.5$$

**If the time step is larger than the value specified by the above eq., the resulted numerical solutions will be oscillating . We call the solution procedure is not stable.**

**More discussion on the numerical stability will be presented in Chapter 7.**

## **3.5 FDHT in Circular Tubes**

**3.5.1 Introduction to FDHT in tubes and ducts**

**3.5.2 Physical and Mathematical Models**

**3.5.3 Governing equations and their non-dimensional forms**

**3.5.4 Conditions for unique solution**

**3.5.5 Numerical solution method**

**3.5.6 Treatment of numerical results**

**3.5.7 Discussion on numerical results**

## 3.5 Fully Developed HT in Circular Tubes

### 3.5.1 Introduction to FDHT in tubes and ducts

#### 1. Simple fully developed heat transfer

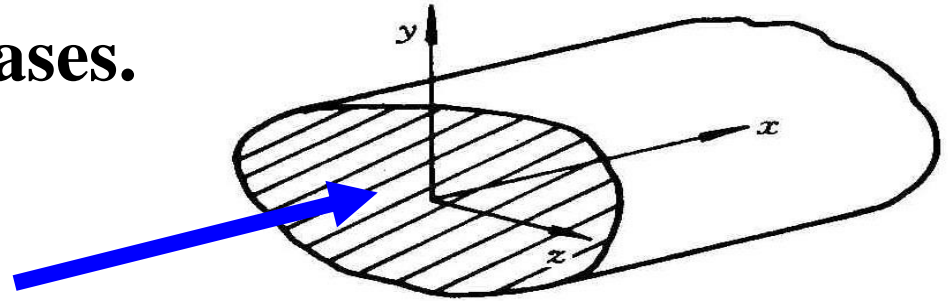
**Physically:** Velocity components normal to flow direction equal zero; Fluid dimensionless temperature distribution is independent on(无关) the position in the flow direction

**Mathematically:** Both dimensionless momentum and energy equations are of **diffusion type**.

Present chapter is limited to simple cases.

**FDHT in straight duct  
is an example of simple cases.**

$$\frac{\partial}{\partial x} \left( \frac{T_{w,m} - T}{T_{w,m} - T_b} \right) = 0$$



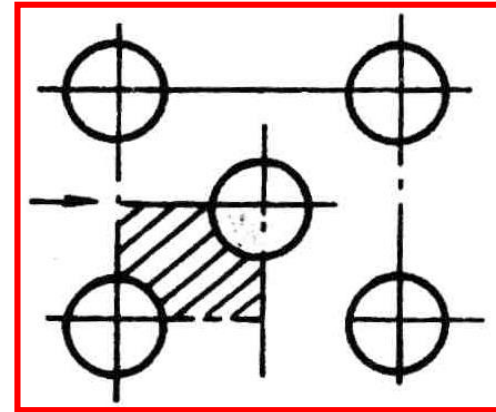
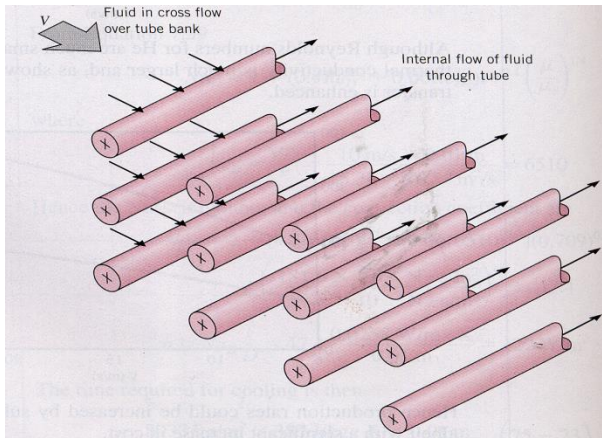
## 2. Complicated FDHT

In the cross section normal to flow direction there exist velocity components, and the dimensionless temperature depends on the axial position, often exhibits periodic (周期的) character. The full Navier-Stokes equations must be solved.

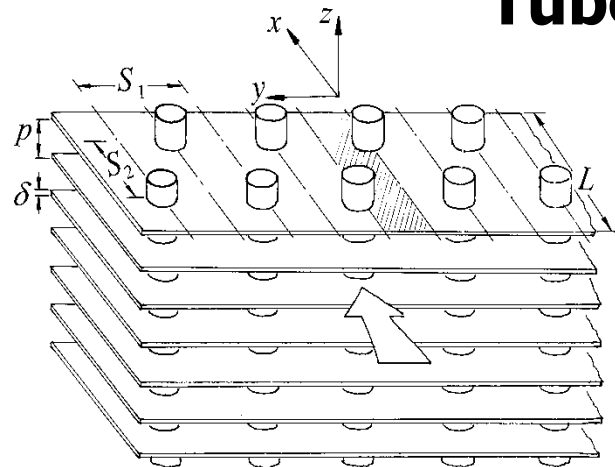
**This subject is discussed in Chapter 11 of the textbook.**



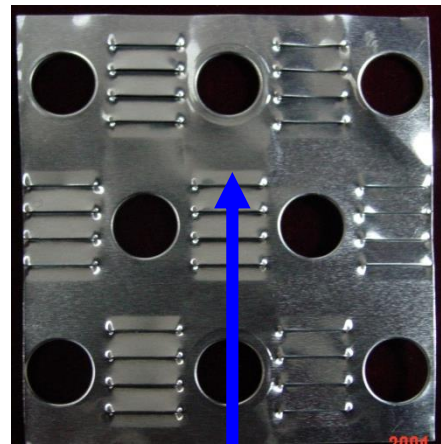
# Examples of complicated FDHT



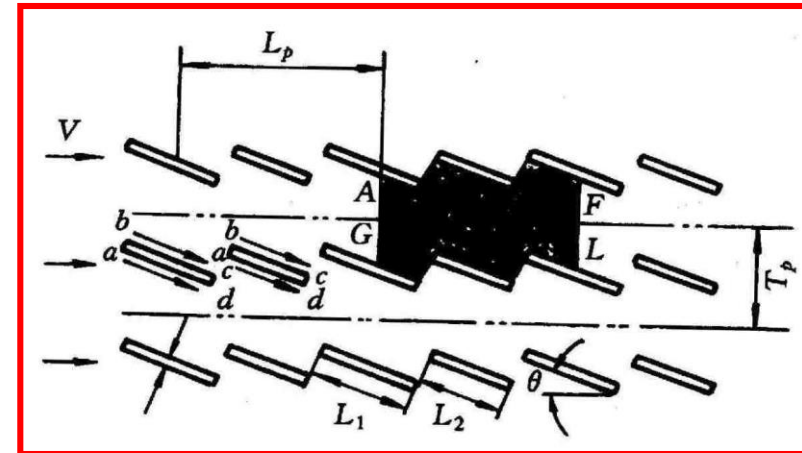
**Tube bundle (bank) (管束)**



**Fin-and-tube heat exchanger**

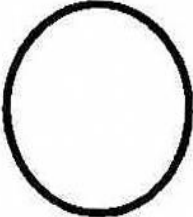




**Louver fin (百叶窗翅片)**



### 3. Collection of partial examples

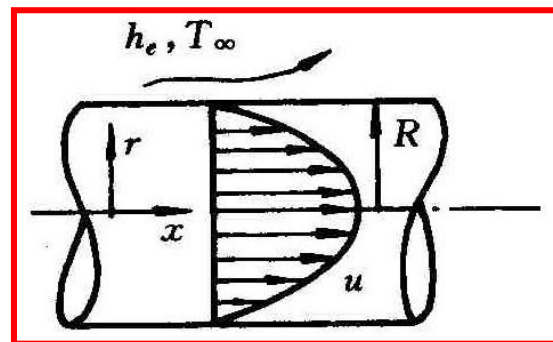
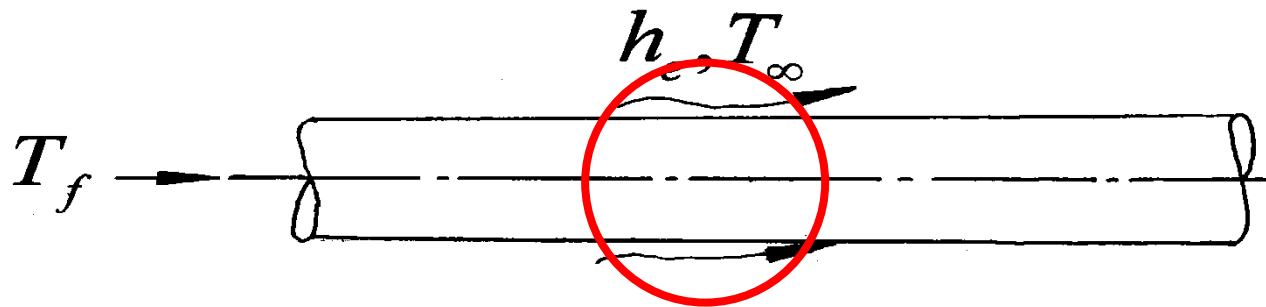
**Table 4-5 Numerical examples of simple FDHT**

No	Cross section	B. Condition	Refs
1		均匀壁温; 给定周向热流分布; 轴向热流呈指数变化; 外部对流换热	[23,24,25,26,27]
2		均匀壁温; 均匀热流及其组合	转引自[23]
3		均匀壁温; 周向任意分布热流; 轴向均匀热流; 一组对边均匀壁温, 另一线绝热	[28,29,30]

**See pp. 106-109 for details**

## 3.5.2 Physical and mathematical models

A laminar flow in a long tube is cooled (heated) by an external fluid with temperature  $T_\infty$  and heat transfer coefficient  $h_e$ . Determine the in-tube heat transfer coefficient and Nusselt number in the FDHT region.



# 1. Simplification (简化) assumptions

- (1) Thermo-physical properties are constant ;
- (2) Axial heat conduction in the fluid is neglected
- (3) Viscous dissipation (耗散) is neglected;
- (4) Natural convection is neglected;
- (5) Wall thermal resistance is neglected;
- (6) The flow is fully developed:

$$\frac{u}{u_m} = 2\left[1 - \left(\frac{r}{R}\right)^2\right]; \quad v = 0$$

## 2. Mathematical formulation (描述)

### (1) Energy equation

Cylindrical coordinate, symmetric temp. distribution, and no natural convection (A4):

$$\rho c_p \left( u \frac{\partial T}{\partial x} + v \frac{\partial T}{\partial r} \right) = \frac{1}{r} \frac{\partial}{\partial r} \left( \lambda r \frac{\partial T}{\partial r} \right) + \frac{\partial}{\partial x} \left( \lambda \frac{\partial T}{\partial x} \right) + S_T$$

**FD flow  
(A6)**

**No axial  
cond.  
(A2)**

**No  
dissipation  
(A3)**

$$\rho c_p u \frac{\partial T}{\partial x} = \frac{1}{r} \frac{\partial}{\partial r} \left( \lambda r \frac{\partial T}{\partial r} \right)$$

**Type of eq.?**

**2-D parabolic eq.!**

## (2) Boundary condition

$$r = 0, \frac{\partial T}{\partial r} = 0 \quad (\text{Symmetric condition}) ;$$

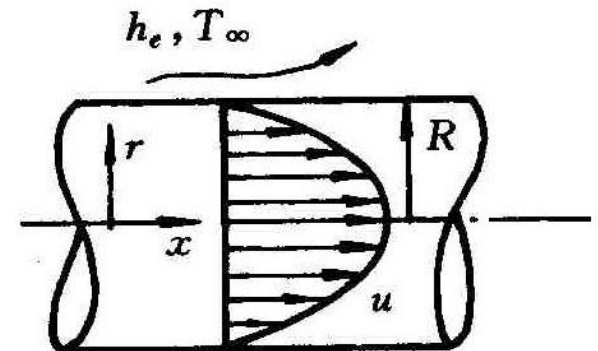
$$r = R, -\lambda \frac{\partial T}{\partial r} = h_e (T - T_\infty)$$

(External convective condition!)

Internal fluid thermal conductivity

External (外部) convective heat transfer

No wall thermal resistance (A5), tube outer radius = R



### 3.5.3 Governing eqs. and dimensionless forms

From fully developed condition a dimensionless temperature can be introduced, transforming the PDE to ordinary eq..

Defining  $\Theta = \frac{T - T_\infty}{T_b - T_\infty}$  ←  $\frac{T - T}{T_b - T}$  ←  $\frac{T - T}{T - T}$

Then:  $T = \Theta(T_b - T_\infty) + T_\infty$ ;  $\frac{\partial T}{\partial x} = \Theta \frac{\partial T_b}{\partial x} = \Theta \frac{dT_b}{dx}$

Defining dimensionless space and “time” coordinates:

$$\eta = \frac{r}{R}; \quad X = \frac{x}{R \bullet Pe} \quad Pe = \frac{2R\rho c_p u_m}{\lambda} = \frac{2Ru_m}{a}$$

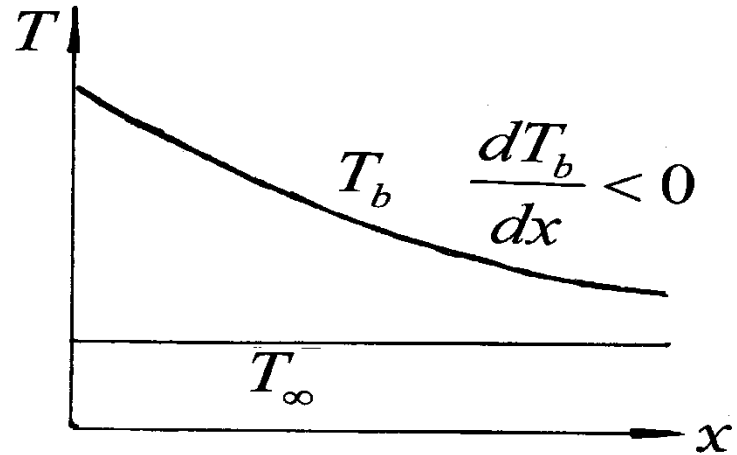
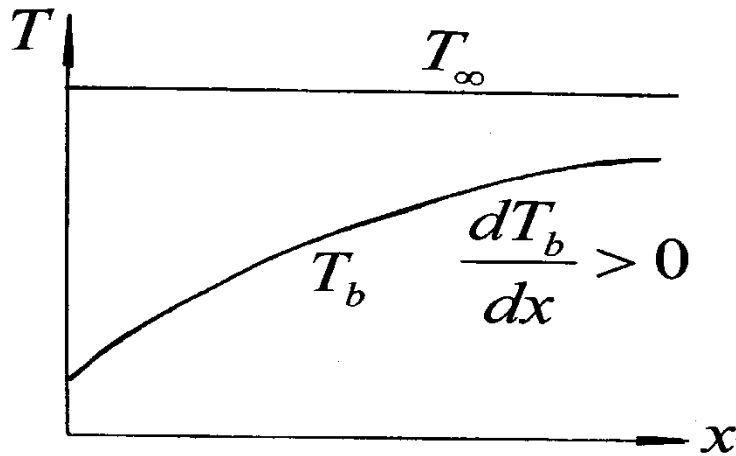
Constant properties (A 1)

Energy eq. can be rewritten as:

$$\frac{dT_b / dX}{T_b - T_\infty} = \frac{1}{\eta} \frac{d}{d\eta} \left( \eta \frac{d\Theta}{d\eta} \right) / \left( \frac{1}{2} \Theta \frac{u}{u_m} \right) = -\Lambda \quad \Lambda > 0$$

Dependent on X only

Dependent on  $\eta$  only



$\Lambda$  is called **eigenvalue** (特征值)



Following ordinary differential equation for the dimensionless temperature can be obtained

$$\frac{1}{\eta} \frac{d}{d\eta} \left( \eta \frac{d\Theta}{d\eta} \right) / \left( \frac{1}{2} \Theta \frac{u}{u_m} \right) = -\Lambda \quad (\text{a})$$

The original two B.Cs. are transformed (转换成) into:

$$\eta = 0, \quad \frac{d\Theta}{d\eta} = 0; \quad (\text{b})$$

$$\eta = 1, \quad -\frac{d\left(\frac{T - T_\infty}{T_b - T_\infty}\right)}{d\left(\frac{r}{R}\right)} = \left(\frac{h_e R}{\lambda}\right) \frac{T - T_\infty}{T_b - T_\infty} \longrightarrow \left(\frac{d\Theta}{d\eta}\right)_{\eta=1} = -Bi\Theta_w \quad (\text{c})$$

**Question:** whether from Eqs. (a)-(c) a unique (唯一的) solution can be obtained?

## 3.5.4 Analysis of condition for unique solution

Because of the **homogeneous** (齐次性) character :

**Every term in the differential equation contains a linear part of dependent variable or its 1<sup>st</sup>/2nd derivative.**

$$\frac{1}{\eta} \frac{d}{d\eta} \left( \eta \frac{d\Theta}{d\eta} \right) / \left( \frac{1}{2} \Theta \frac{u}{u_m} \right) = -\Lambda \longrightarrow \frac{1}{\eta} \frac{d}{d\eta} \left( \eta \frac{d\Theta}{d\eta} \right) = -\Lambda \left( \frac{1}{2} \Theta \frac{u}{u_m} \right)$$

In addition, the given B.Cs. are also **homogeneous**:

$$\eta = 0, \frac{d\Theta}{d\eta} = 0; \quad \left. \frac{d\Theta}{d\eta} \right|_{\eta=1} = -Bi\Theta_w$$

For the above mathematical formulation there exists an uncertainty (不确定性) of being able to be multiplied by a constant for its solution.

While in order to solve the problem, the value of  $\Lambda$  in the formulation has to be determined.

**In order to get a unique solution and to specify the eigenvalue, we need to supply one more condition!**

**We examine the definition of dimensionless temperature:**

$$\Theta_b = \left( \frac{T - T_\infty}{T_b - T_\infty} \right)_b = \frac{T_b - T_\infty}{T_b - T_\infty} \equiv \mathbf{1.0}$$

**Physically, the averaged temp. is defined by**

$$\Theta_b = \frac{\int_0^R 2\pi r u \Theta dr}{\pi R^2 u_m} = 2 \int_0^1 \frac{r}{R} \frac{u}{u_m} \Theta d\left(\frac{r}{R}\right) = \mathbf{1}$$

Thus the complete formulation is:

$$\frac{1}{\eta} \frac{d}{d\eta} \left( \eta \frac{d\Theta}{d\eta} \right) + \Lambda \left( \frac{1}{2} \Theta \frac{u}{u_m} \right) = 0 \quad (\text{a})$$

$$\eta = 0, \quad \frac{d\Theta}{d\eta} = 0; \quad (\text{b})$$

$$\left. \frac{d\Theta}{d\eta} \right)_{\eta=1} = -Bi\Theta_w \quad (\text{c})$$

$$\int_0^1 \eta \frac{u}{u_m} \Theta d\eta = 1/2 \quad (\text{d})$$

**Non-homogeneous term!**

## 3.5.5 Numerical solution method

$$\frac{1}{\eta} \frac{d}{d\eta} \left( \eta \frac{d\Theta}{d\eta} \right) + \Lambda \left( \frac{1}{2} \Theta \frac{u}{u_m} \right) = 0$$

**This is a 1-D conduction equation with a source term!**

$\frac{\Lambda}{2} \Theta \frac{u}{u_m}$ , whose value should be determined during the solution process **iteratively (迭代地)**.

**Patankar – Sparrow** proposed following numerical solution method:

**(1) Let  $\Theta = \Lambda \phi$**

Because of the homogeneous character, the form of the equation is not changed only replacing  $\Theta$  by  $\phi$ .

$$\frac{1}{\eta} \frac{d}{d\eta} \left( \eta \frac{d\phi}{d\eta} \right) + \Lambda \left( \frac{1}{2} \phi \frac{u}{u_m} \right) = 0 \quad \text{(a)}$$

$$\eta = 0, \quad \frac{d\phi}{d\eta} = 0; \quad \text{(b)}$$

$$\left. \frac{d\phi}{d\eta} \right|_{\eta=1} = -Bi\phi_w \quad \text{(c)}$$

$$\int_0^1 \eta \frac{u}{u_m} \Lambda \phi d\eta = 1/2 \quad \text{(d)} \quad \longrightarrow$$

**Non-homogeneous term**

$$\Lambda = 1 / \left( 2 \int_0^1 \eta \frac{u}{u_m} \phi d\eta \right) \quad \text{It can be used to iteratively}$$

**determine the eigenvalue.**

**(2) Assuming an initial field  $\phi^*$ , get  $\Lambda^*$**

**(3) Solving an ordinary differential eq. with a source term to get an improved  $\phi$**

**(4) Repeating the above procedure until**

$$\left| \frac{\phi^* - \phi}{\phi} \right| \leq \varepsilon, \quad \varepsilon = 10^{-3} \sim 10^{-6}$$

**This iterative procedure is easy to approach convergence:**

$$S = \Lambda \frac{1}{2} \frac{u}{u_m} \phi = \frac{(u/u_m)\phi}{4 \int_0^1 \eta (u/u_m) \phi d\eta} = \frac{(1-\eta^2)\phi}{4 \int_0^1 \eta (1-\eta^2) \phi d\eta}$$

$$\Lambda = 1 / \left( 2 \int_0^1 \eta \frac{u}{u_m} \phi d\eta \right)$$

# Brief review of 2017-09-19 lecture key points

## 1. Linearization of source

$$S = S_C + S_P \phi_P, \quad S_P \leq 0$$

$S_P \leq 0$  is needed to ensure the sufficient condition for the solution convergence of the algebraic equations.

## 2. Additional source term method (ASTM) for treating 2<sup>nd</sup> and 3<sup>rd</sup> kind boundary conditions

Regarding the heat going into the solution domain by 2<sup>nd</sup> or 3<sup>rd</sup> kind B.C. as the **source term** of the first inner control volume;

Cutting the connection between inner node and boundary to close the algebraic eqs of the inner nodes by eliminating the unknown wall temp. from them.



### 3. TDMA solution algorithm for 1-D problem

(1) Elimination (消元) – Reducing the unknowns at each line from 3 to 2

$$A_i T_i = B_i T_{i+1} + C_i T_{i-1} + D_i \text{(a)} \rightarrow T_{i-1} = P_{i-1} T_i + Q_{i-1} \text{(b)}$$

$$P_i = \frac{B_i}{A_i - C_i P_{i-1}}; Q_i = \frac{D_i + C_i Q_{i-1}}{A_i - C_i P_{i-1}}; \leftarrow P_1 = \frac{B_1}{A_1}; Q_1 = \frac{D_1}{A_1}$$

recursive

(2) Back substitution(回代) – Starting from the last node via Eq.(b) to get  $T_i$  sequentially

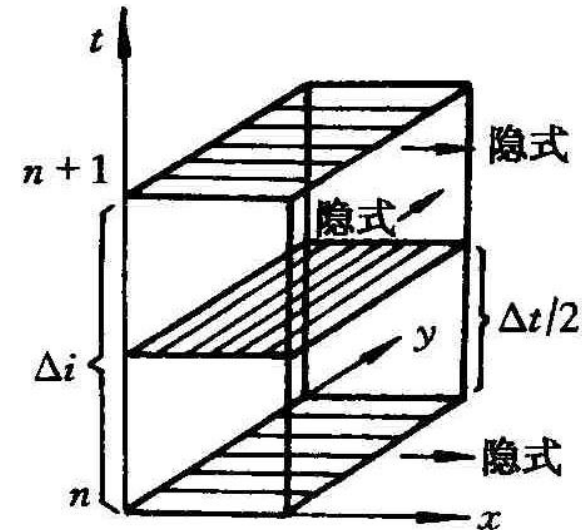
### 4. ADI method for solving 2-D unsteady problem

Dividing  $\Delta t$  into two uniform parts; In the 1<sup>st</sup>  $\Delta t / 2$

implicit in  $x$  direction, and explicit in  $y$  direction;

In the 2<sup>nd</sup>  $\Delta t / 2$  implicit in  $y$  direction, and explicit in  $x$  direction.

By implementing TDMA twice, the algebraic equations for forwarding one time step is solved.



## 5. Homogeneous problems

Every term in the differential equation and boundary conditions only contains a linear part of dependent variable or its 1<sup>st</sup> or 2<sup>nd</sup> derivative.

For such a mathematical formulation there exists an uncertainty of being able to be multiplied by a constant for its solution.

$\phi$  exists in both numerator and denominator, thus only the distribution, rather than absolute value will affect the source term. (20180919)

### 3.5.6 Treatment of numerical results

Two ways for obtaining heat transfer coefficient:

1. From solved temp. distribution using Fourier's law of heat conduction and Newton's law of cooling:

$$r = R, -\lambda \frac{\partial T}{\partial r} = h(T_w - T_b) \rightarrow h = -\lambda \left( \frac{\partial T}{\partial r} \right)_{r=R} \frac{1}{T_w - T_b}$$

**For inner fluid**

**Note: different from Boundary condition**

$$r = R, -\lambda \frac{\partial T}{\partial r} = h_e (T - T_\infty)$$

## 2. From the eigenvalue (特征值) :

From heat balance between inner and external heat transfer

$$h(T_b - T_w) = h_e(T_w - T_\infty)$$

Inner

External

Get:

$$\begin{aligned}
 h = h_e \frac{T_w - T_\infty}{T_b - T_w} &\rightarrow h = h_e \frac{1}{\frac{T_b - T_w}{T_w - T_\infty}} \rightarrow \frac{h_e}{\frac{T_b - T_\infty + T_\infty - T_w}{T_w - T_\infty}} \\
 \rightarrow \frac{h_e}{\frac{T_b - T_\infty}{T_w - T_\infty} - 1} &\rightarrow h = \frac{h_e}{\frac{1}{\frac{T_w - T_\infty}{T_b - T_\infty}} - 1} = \frac{h_e}{\frac{1}{\Theta_w} - 1}
 \end{aligned}$$

$$h = \frac{h_e}{\frac{1}{\Theta_w} - 1} = \frac{h_e \Theta_w}{1 - \Theta_w} = \frac{h_e \Lambda \phi_w}{1 - \Lambda \phi_w}$$

$$Nu = \frac{2Rh}{\lambda} = \frac{2R}{\lambda} \frac{h_e \Lambda \phi_w}{1 - \Lambda \phi_w} = \frac{2Bi \Lambda \phi_w}{1 - \Lambda \phi_w}$$

From the specified values  $Bi$ , the corresponding eigenvalues,  $\Lambda$ , can be obtained. Thus it is not necessary to find the 1<sup>st</sup>-order derivative at the wall of function  $\phi$  for determining Nusselt number.

### 3.5.7 Discussion on numerical results

**Table : Numerical results of FDHT in tubes**  
**In the textbook: Table 4-6**

$Bi$	$\Lambda$	$Nu$
0	0	4.364
0.1	0.381 8	4.330
0.25	0.894 3	4.284
0.5	1.615	4.221
1	2.690	4.122
2	3.995	3.997
5	5.547	3.840
10	6.326	3.758
100	7.195	3.663
$\infty$	7.314	3.657

$(Nu)_q$   
 $(Nu)_T$

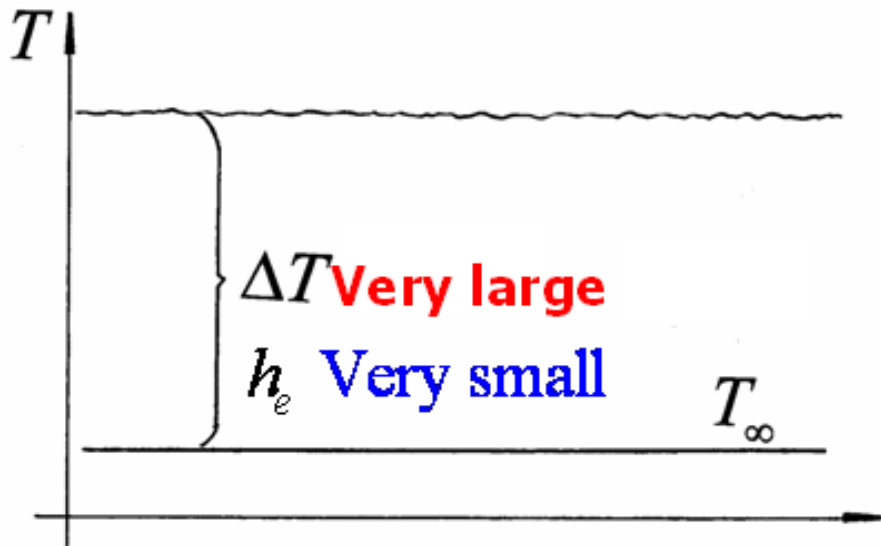
**1.  $Bi$  effect:**

From definition  $Bi = \frac{Rh_e}{\lambda}$

$Bi \rightarrow \infty, h_e \rightarrow \infty$  External heat transfer is very strong, the wall temp. approaches fluid temp.

This is corresponding to constant wall temp condition,  
Thus  **$Nu = 3.66$**

$Bi \rightarrow 0, h_e \rightarrow 0$  **Is this adiabatic? No!**



Product of very small HT coefficient and very large temp. difference makes heat flux almost constant.

$q = h_e \Delta T \approx const$

## 2. Computer implementation of $Bi \rightarrow \infty$ and $Bi = 0$

**$Bi \rightarrow \infty$  by progressively (逐渐地) increasing  $Bi$ :**

$$Bi = 10^5, 10^6, 10^7, \dots$$

**$Bi = 0$  by progressively decreasing  $Bi$ :**

$$Bi = 0.1, 0.01, 0.001, 0.0001, 0.00001,$$

**Double decision (双精度) must be used for Computation:**

$$Nu = \frac{2Bi\Lambda\phi_w}{1 - \Lambda\phi_w}, \quad Bi \rightarrow 0, \quad \Lambda \rightarrow 0, \quad \Lambda\phi_w \rightarrow 1 \rightarrow \frac{0}{0}$$



## **4.6 Fully Developed HT in Rectangle Ducts**

### **4.6.1 Physical and mathematical models**

### **4.6.2 Governing eqs. and their dimensionless forms**

### **4.6.3 Condition for unique solution**

### **4.6.4 Treatment of numerical results**

### **4.6.5 Other cases**

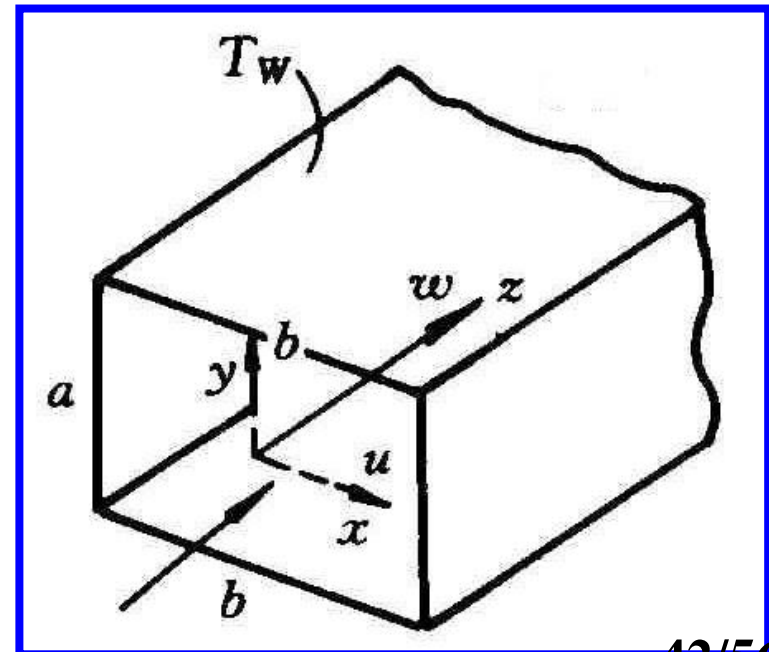
## 3.6 Fully Developed HT in Rectangle Ducts

### 3.6.1 Physical and mathematical models

Fluid with constant properties flows in a long rectangle duct with a constant wall temp. **Determine the friction factor and HT coefficient in the fully developed region for laminar flow.**

#### 1. Momentum eq.

For the fully developed flow  $u=v=0$ , only the velocity component in z-direction is not zero. Its governing equation:



$$\rho \left( u \frac{\partial w}{\partial x} + v \frac{\partial w}{\partial y} + w \frac{\partial w}{\partial z} \right) = -\frac{\partial p}{\partial z} + \eta \left( \frac{\partial^2 w}{\partial x^2} + \frac{\partial^2 w}{\partial y^2} + \frac{\partial^2 w}{\partial z^2} \right)$$

**Neglecting cross section variation of  $p$**

$$\eta \left( \frac{\partial^2 w}{\partial x^2} + \frac{\partial^2 w}{\partial y^2} \right) - \frac{\partial p}{\partial z} = 0 \qquad \eta \left( \frac{\partial^2 w}{\partial x^2} + \frac{\partial^2 w}{\partial y^2} \right) - \frac{dp}{dz} = 0$$

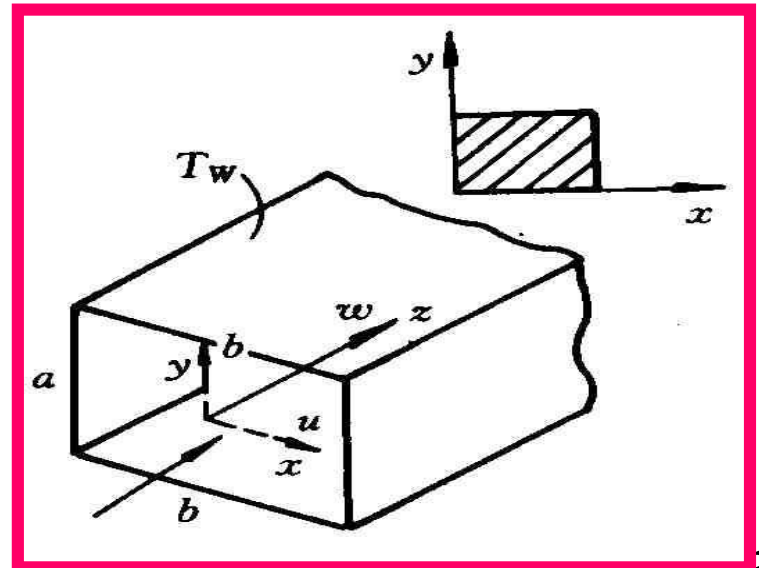
Taking 1/4 region as the computational domain because of symmetry. Boundary conditions are:

At the wall,  $w=0$ ;

At center line,

First order normal derivative equals zero:

$$\frac{\partial w}{\partial n} = 0$$



Defining a dimensionless velocity as :

$$W = \frac{\eta w}{-D^2 \frac{dp}{dz}}$$

where  $D$  is the referenced length, say:  $D = a$ , or  $D = b$ .

Defining dimensionless coordinates:  $X = x/D$ ,  $Y = y/D$ , then:

$$\eta \left( \frac{\partial^2 w}{\partial x^2} + \frac{\partial^2 w}{\partial y^2} \right) - \frac{dp}{dz} = 0 \rightarrow \left\{ \begin{array}{l} \frac{\partial^2 W}{\partial X^2} + \frac{\partial^2 W}{\partial Y^2} + 1 = 0 \\ \text{At wall, } W = 0; \\ \text{At center lines, } \frac{\partial W}{\partial n} = 0 \end{array} \right.$$

It is a heat conduction problem with a source

term and a constant diffusivity  $\eta$  !

## 2. Energy equation

$$\rho c_p \left( \cancel{u \frac{\partial T}{\partial x}} + \cancel{v \frac{\partial T}{\partial y}} + w \frac{\partial T}{\partial z} \right) = \frac{\partial}{\partial x} \left( \lambda \frac{\partial T}{\partial x} \right) + \frac{\partial}{\partial y} \left( \lambda \frac{\partial T}{\partial y} \right) + \frac{\partial}{\partial z} \left( \lambda \cancel{\frac{\partial T}{\partial z}} \right)$$

Thus: 
$$\rho c_p w \frac{\partial T}{\partial z} = \frac{\partial}{\partial x} \left( \lambda \frac{\partial T}{\partial x} \right) + \frac{\partial}{\partial y} \left( \lambda \frac{\partial T}{\partial y} \right)$$

**Neglecting axial  
heat conduction**

**Type of equation? Parabolic! Z is a one-way  
coordinate like time!**

**Boundary conditions:**

At the wall,  $T = T_w$  ;

At the center line,  $\frac{\partial T}{\partial n} = 0$

## 3.6.2 Dimensionless governing equation

We should define an appropriate dimensionless temperature such that the dimension of the problem can be reduced from 3 to 2: **Separating the one-way coordinate  $z$  from the two-way coordinates  $x, y$ .**

$$\Theta = \frac{T_w - T}{T_w - T_b} \quad \leftarrow \quad \frac{T - T_b}{T_w - T_b} \quad \leftarrow \quad \frac{T - T_b}{T_w - T_b}$$

Then  $T = \Theta(T_b - T_w) + T_w$

$$\frac{\partial T}{\partial z} = \Theta \frac{\partial (T_b - T_w)}{\partial z}$$

$$Pe = \frac{\rho c_p w_m D}{\lambda}$$

Defining:  $X = x/D, Y = y/D, Z = z/(DPe)$

**One-way coordinate!**

**Dimensionless governing eq.**

$$\frac{\partial(T_b - T_w)}{\partial Z} \frac{1}{T_b - T_w} = \frac{\frac{\partial^2 \Theta}{\partial X^2} + \frac{\partial^2 \Theta}{\partial Y^2}}{\frac{W}{W_m} \Theta} = -\Lambda$$

$\Lambda > 0$

**Dependent on Z only**

**Dependent on X, Y only**

**Thus:**

$$\frac{\partial^2 \Theta}{\partial X^2} + \frac{\partial^2 \Theta}{\partial Y^2} + \Lambda \frac{W}{W_m} \Theta = 0;$$

$$\frac{d(T_b - T_w)}{dZ} \frac{1}{T_b - T_w} = -\Lambda$$

**At the wall**  $\Theta = 0$

**At center line,**  $\frac{\partial \Theta}{\partial n} = 0$

**Heat conduction with an inner source!**

### 3.6.3 Analysis on the unique solution condition

Because of the homogeneous character, these also exists an uncertainty of being magnifying by any times!

Introducing average temperature (difference):

$$T_w - T_b = \frac{\int (T_w - T) w dA}{\int w dA} \longrightarrow \frac{T_w - T_b}{T_w - T_b} = \frac{\int \frac{T_w - T}{T_w - T_b} w dA}{w_m A}$$

$$1 = \frac{1}{A} \int \frac{T_w - T}{T_w - T_b} \frac{w}{w_m} dA \longrightarrow 1 = \frac{1}{A} \int \Theta \left( \frac{W}{W_m} \right) dA$$

**It is the additional condition for the unique solution.**

Numerical solution method is the same as that for a circular tube.



## 3.6.4 Treatment of numerical results\*

After receiving converged velocity and temperature fields, friction factor and Nusselt number can be obtained as follows:

**1.  $fRe$ — for laminar problems  $fRe = \text{constant}$ :**

$$f Re = \left[ -\frac{D_e \frac{dp}{dz}}{\frac{1}{2} \rho w_m^2} \right] \left( \frac{w_m D_e}{\nu} \right) \xrightarrow{\text{Definition of } W} f Re = \frac{2}{W_m} \left( \frac{D_e}{D} \right)^2$$

$$W = \frac{\eta w}{-D^2 \frac{dp}{dz}}$$

**2.  $Nu$ — Making an energy balance :**

$$\rho c_p w_m A \frac{dT_b}{dz} = qP, P \text{ is the duct circumference length}$$

$$\frac{d(T_b - T_w)}{dZ} \frac{1}{T_b - T_w} = -\Lambda \quad \text{i.e.,} \quad \frac{dT_b}{dZ} = \frac{dT_b}{dz} DPe = (T_w - T_b)\Lambda$$

$$\frac{dT_b}{dz} = \frac{1}{DPe} (T_w - T_b)\Lambda \quad \text{Substituting in}$$

$$\rho c_p w_m A \frac{dT_b}{dz} = qP$$

yields  $q = \frac{A \rho c_p w_m}{P} \frac{dT_b}{dz} = \frac{A \rho c_p w_m}{P} \frac{1}{DPe} \Lambda (T_w - T_b)$

yields:  $q = \frac{A \lambda}{P D^2} \Lambda (T_w - T_b)$

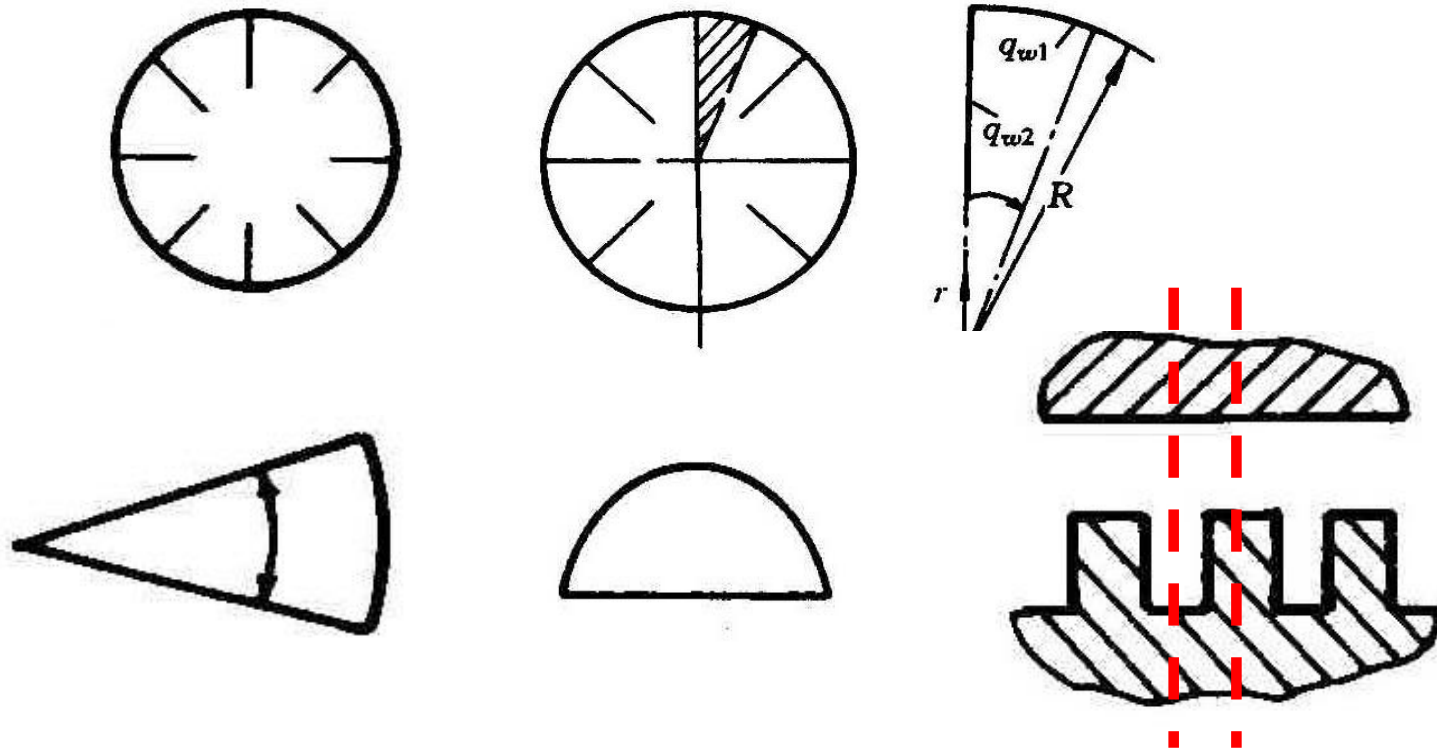
$$Pe = \frac{\rho c_p w_m D}{\lambda}$$

$$Nu = \frac{hD_e}{\lambda} = \frac{q}{T_w - T_b} \frac{D_e}{\lambda} = \frac{1}{T_w - T_b} \frac{D_e}{\lambda} \frac{A \lambda}{P D^2} \Lambda (T_w - T_b) = \frac{1}{4} \left(\frac{D_e}{D}\right)^2 \Lambda$$

$$D_e = \frac{4A}{P}$$

$$f \text{Re} = \frac{2}{W_m} \left(\frac{D_e}{D}\right)^2 \quad Nu = \frac{1}{4} \left(\frac{D_e}{D}\right)^2 \Lambda$$

### 3.6.5 Other cases



## Home Work 3

4-2 ( $T_1=150, T_f=25$ ),

4-4,

4-12,

4-14,

4-18

**Due in October 8**

**Problem 4-2:** As shown in Fig. 4-22, in 1-D steady heat conduction problem, known conditions are:  $T_1=150$ ,  $\lambda=5$ ,  $S=150$ ,  $T_f=25$ ,  $h=15$ , the units in every term are consistent. Try to determine the values of  $T_2, T_3$ ; Prove that the solution meet the overall conservation requirement even though only three nodes are used.

**Problem 4-4:** A large plate with thickness of 0.1 m, uniform source  $S=50 \times 10^3 \text{ W/m}^3$ ,  $\lambda = 10 \text{ W} / (\text{m} \cdot ^\circ \text{C})$  ; One of its wall is kept at  $75^\circ \text{C}$ , while the other wall is cooled by a fluid with  $T_f = 25^\circ \text{C}$  and heat transfer coefficient  $h = 50 \text{ W/m}^2 \cdot ^\circ \text{C}$

Adopt Practice B, divide the plate thickness into three uniform CVs, determine the inner node temperature. Take 2<sup>nd</sup> order accuracy for the inner node, adopt the additional source term method for the right boundary node.

## Problem 4-12:

Write a program using TDMA algorithm, and use the following method to check its accuracy: set arbitrary values of the coefficients  $A_i, B_i$  and  $C_i$  ( $i = 1, 10$ ). But  $B_1$  and  $C_{10}$  should not be zero. Then setting the reasonable values of temperature  $T_1, \dots, T_{10}$ , calculate the corresponding constants  $D_i$ . Apply your program for solving  $T_i$  by using the values of  $A_i, B_i, C_i$  and  $D_i$ , and compare the results with the given value.

## Problem 4-14:

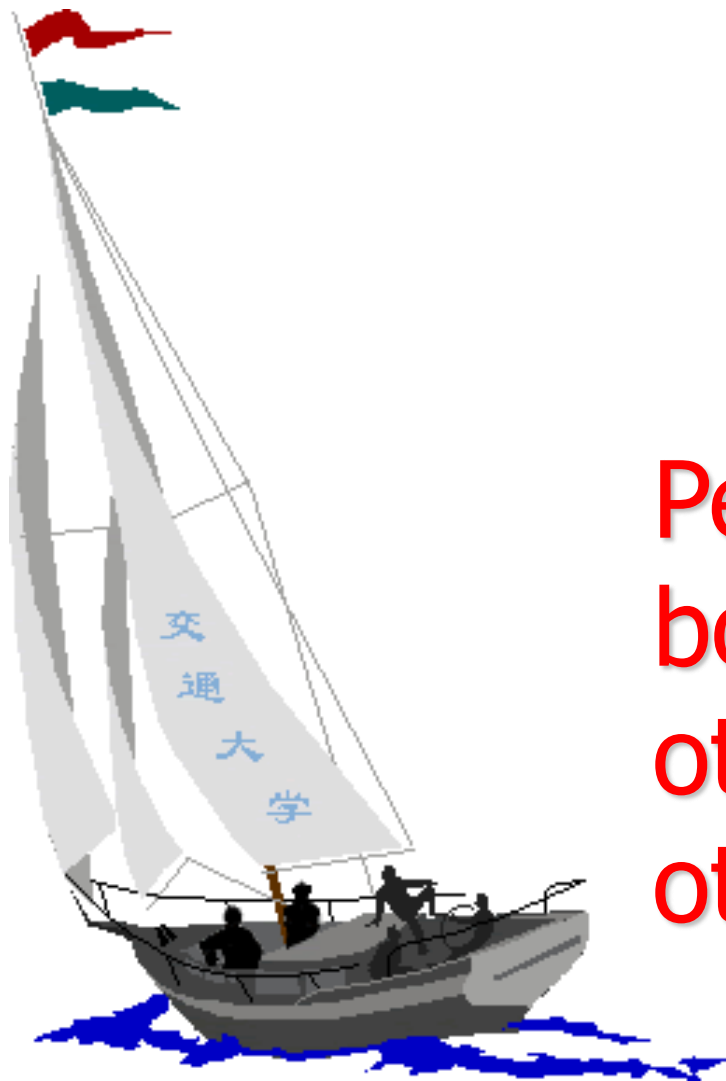
According to the problem discussed in section 4.6 (The fully developed heat convection in a circular tube), try to analyze the following three dimensionless temperature definitions

of  $\Theta = \frac{T - T_w}{T_b - T_w}$ ,  $\Theta = \frac{T - T_\infty}{T_w - T_\infty}$  and  $\Theta = \frac{T - T_w}{T_\infty - T_w}$ , which one is acceptable for separation of

variables.

**Problem 4-18:** Shown in Fig.4-25 is a laminar fully developed heat transfer in a duct of half circular cross. Try:

- (1) Write the mathematical formulation of the heat transfer problem;**
- (2) Make the formulation dimensionless by introducing some dimensionless parameters;**
- (3) Derive the expressions for  $fRe$  and  $Nu$  from numerical solutions, where the characteristic length for  $Re$  and  $Nu$  is the equivalent diameter  $D_e$ .**



# 同舟共济 渡彼岸!

People in the same  
boat help each  
other to cross to the  
other bank, where....