

数值传热学

第六章 求解椭圆型流动与换热的原始变量法 (3)



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Numerical Heat Transfer

(数值传热学)

Chapter 6 Primitive Variable Methods for Elliptic Flow and Heat Transfer (3)



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Xi'an, 2016-Nov.-07

- Chapter 6 Primitive Variable Methods for Elliptic Flow and Heat Transfer

- 6.1 Source terms in momentum equations and two key issues in numerically solving momentum equation
- 6.2 Staggered grid system and discretization of momentum equation
- 6.3 Pressure correction methods for N-S equation
- 6.4 Approximations in SIMPLE algorithm
- 6.5 Discussion on SIMPLE algorithm and criteria for convergence
- 6.6 Developments of SIMPLE algorithm
- 6.7 Methods for accelerating convergence of pressure - correction methods
- 6.8 Boundary condition treatments for open system
- 6.9 Fluid flow & heat transfer in a closed system
- 6.10 Introduction to collocated grid system

6.8 Boundary condition treatments for open system

6.8.1 Selections for outlet boundary

6.4.2 Treatment of outlet boundary with no-recirculation

1. Local one-way
2. Fully developed

6.8.3 Treatment of outlet boundary with recirculation

1. Example with recirculation
2. Suggestion

6.8.4 Methods for outlet normal velocity satisfying total mass conservation

1. Two cases;
2. Application

6.8 Boundary condition treatments for open system

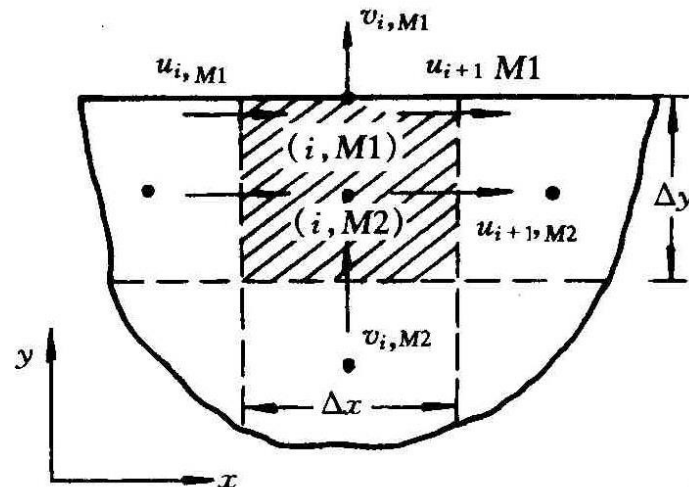
6.8.1 Selections for outlet boundary

1. At the location without recirculation(回流) — Suggested by Patankar
2. At the location with recirculation---special attention should be paid for boundary condition treatment

6.8.2 Treatment of B.C. without recirculation

1. Local one-way

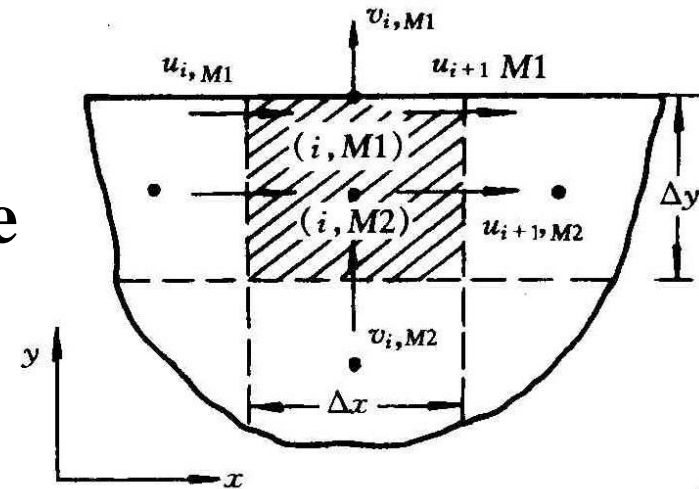
$$(a_N)_{i,M2} = 0$$



2. Fully developed $\left(\frac{\partial \phi}{\partial n}\right)_{i,M1} = 0$

(1) Updating (更新) boundary value

$$\frac{\phi_{i,M1} - \phi_{i,M2}}{(\delta y)_B} = 0 \quad \longrightarrow \quad \phi_{i,M1} = \phi_{i,M2}^*$$



(2) ASTM:

Taking $\left(\frac{\partial \phi}{\partial n}\right)_{i,M1} = 0$ as given heat flux condition

For both methods, **the outlet normal velocity must satisfy total mass conservation condition.**

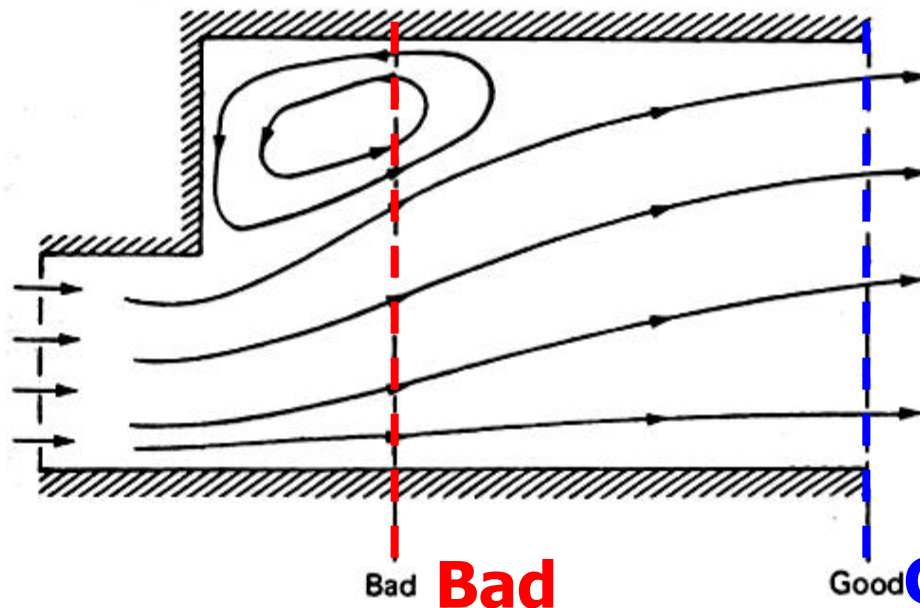
6.8.3 Treatment of outlet boundary with recirculation

1. Necessity for such selection

Required from practical problems.

According to Patankar, the outlet boundary of the sudden expansion case must be positioned at the location without recirculation (“Good” position) . It should not be positioned at the “Bad” location, otherwise the results are meaningless.

This suggestion not only needs more computer memory but also is not possible for some situation.



Open flow system

If the neglect of the diffusion at an outflow boundary appears, for some reason, to be serious, then we should conclude that the analyst has placed the outflow boundary at an inappropriate location. A repositioning of the boundary would normally make the outflow treatment acceptable. A particularly bad choice of an outflow-boundary location is the one in which there is an “inflow” over a part of it. An example of this is shown in Fig. 5.12. For such a bad choice of the boundary, no meaningful solution can be obtained.

This may be a convenient place to review the boundary-condition practices for convection-diffusion problems. Whenever there is no fluid flow across the boundary of the calculation domain, the boundary flux is purely a diffusion flux, and the practices described in Chapter 4 apply. For those parts of the boundary where the fluid flows *into* the domain, usually the values of ϕ are known. (The problem is not properly specified if we do not know the value of ϕ that a fluid stream brings with it.) The parts of the boundary where the fluid *leaves* the calculation domain form the outflow boundary, which we have already discussed.

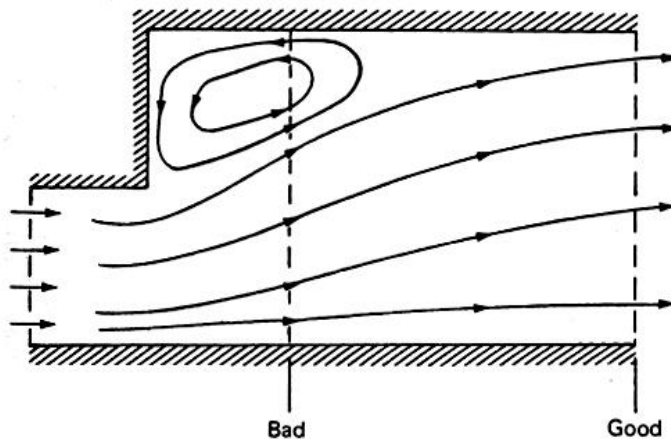
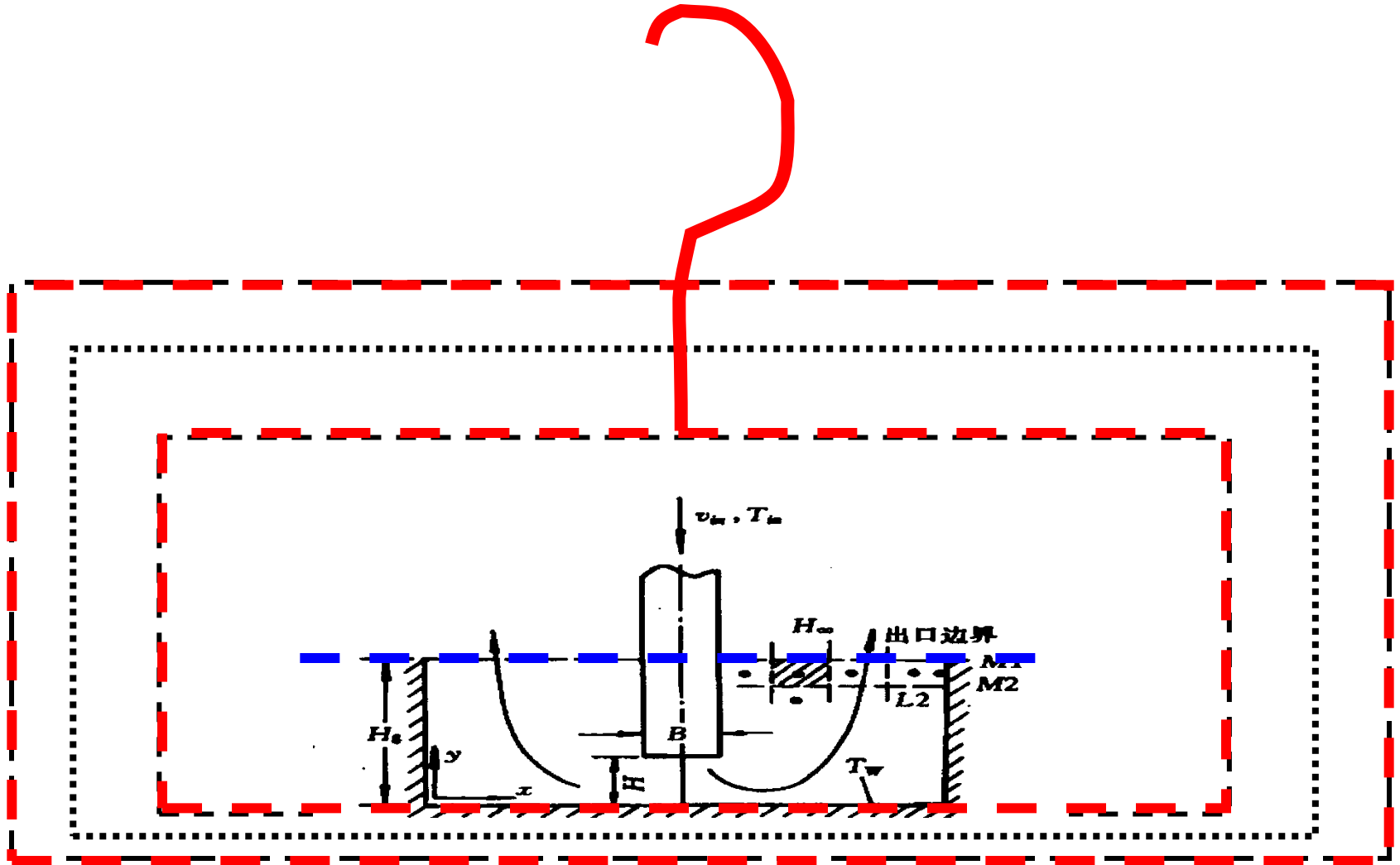


Figure 5.12 Good and bad choices of the location of the outflow boundary.

A particular bad choice of an outflow-boundary location is the one in which there is an “inflow” over a part of it. ... For such a bad choice of the boundary, no meaningful solution can be obtained.

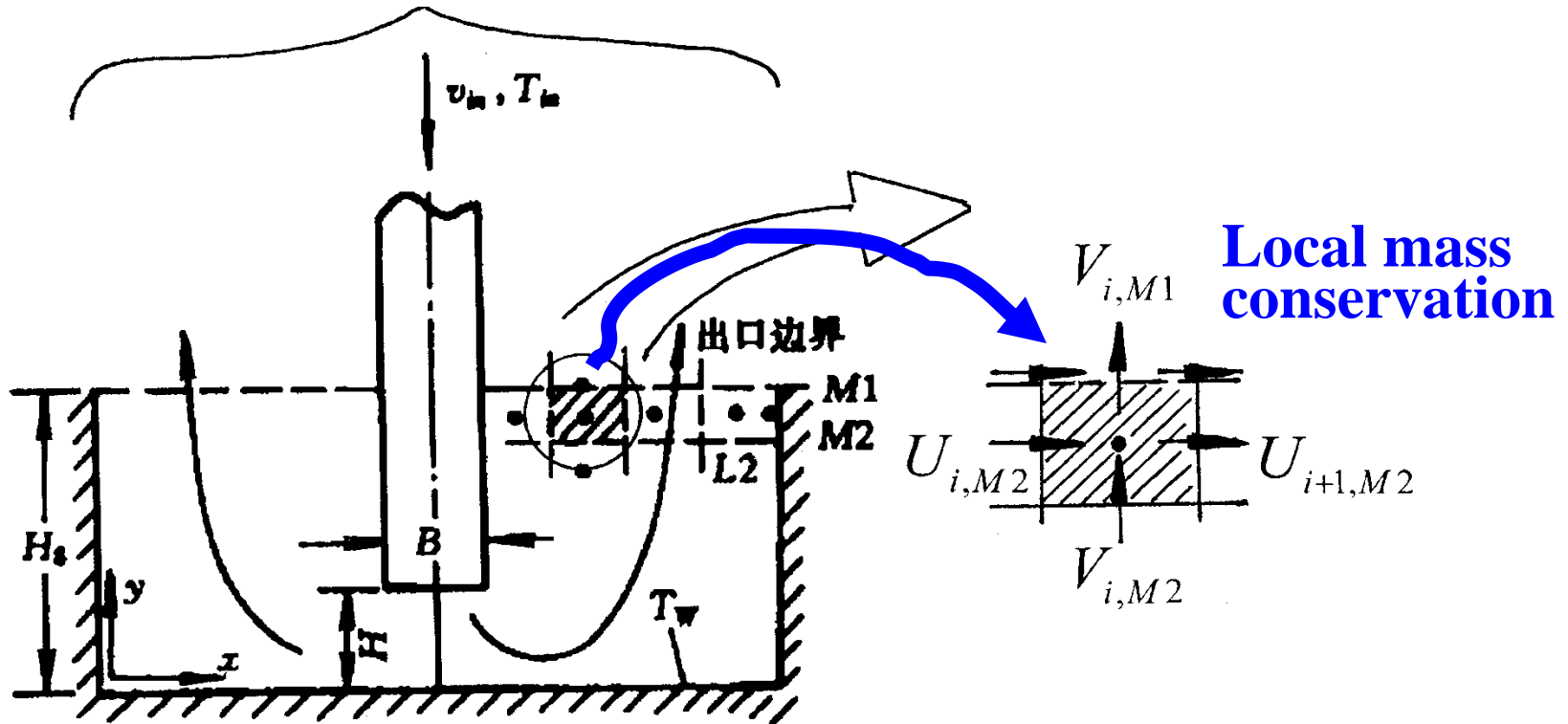


Cooling of plate TV screen

2. Suggestion

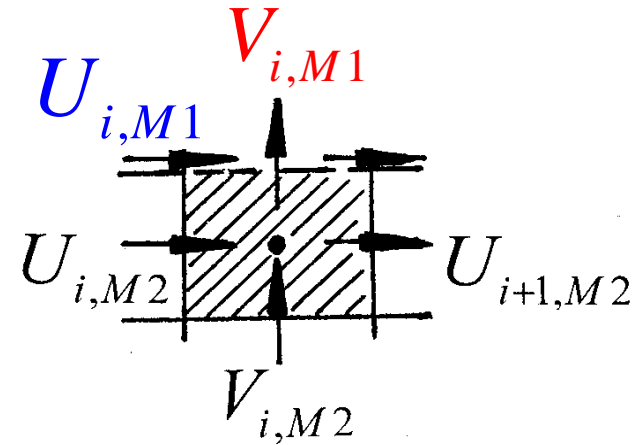
- (1) Outlet normal velocity---treated according to local mass conservation
- (2) Outlet parallel velocity---treated by homogeneous Neumann condition

Total mass conservation



$$\frac{v_{i,M1} - v_{i,M2}}{\Delta y} + \frac{u_{i+1,M2} - u_{i,M2}}{\Delta x} = 0 \rightarrow$$

$$v_{i,M1} = v_{i,M2}^* + \frac{\Delta y}{\Delta x} (u_{i+1,M2}^* - u_{i,M2}^*)$$



The resulted $v_{i,M1}$ has to be corrected by total mass conservation condition.

Tangential velocity

$$\left(\frac{\partial U}{\partial y} \right)_{i,M1} = 0 \rightarrow U_{i,M1} = U_{i,M2}^*$$

6.8.4 Methods for outlet normal velocity satisfying total mass conservation

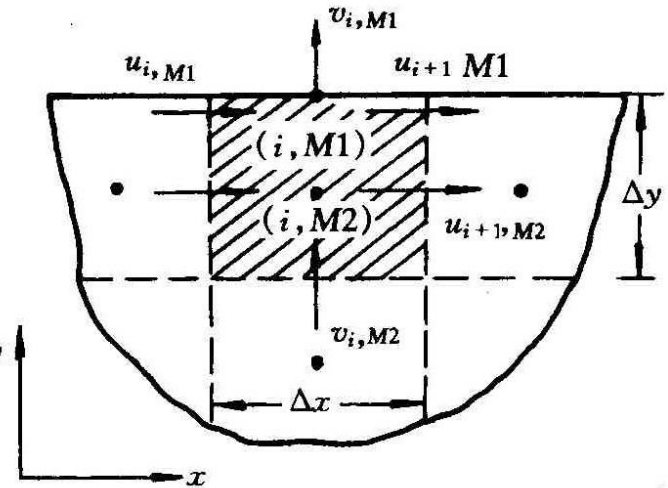
1. Two situations

1) Outlet without recirculation

(1) Relative changes of outlet normal velocity = constant

$$\frac{v_{i,M1} - v_{i,M2}}{v_{i,M2}} = k = \text{const}$$

$$v_{i,M1} = v_{i,M2}(1 + k) = f v_{i,M2}$$



f is determined according to total mass conservation :

$$\sum_{i=2}^{L2} \rho_{i,M1} v_{i,M1} \Delta x_i = \sum_{i=2}^{L2} \rho_{i,M1} f v_{i,M2} \Delta x_i = \text{FLOWIN}$$

$$f = \frac{\text{FLOWIN}}{\sum_{i=2}^{L2} \rho_{i,M1} v_{i,M2} \Delta x_i}$$

$$v_{i,M1} = f \bullet v_{i,M2}^*$$

It is taken as the boundary condition for next iteration.

(2) The 1st derivatives at outlet = constant

$$\frac{v_{i,M1} - v_{i,M2}}{\Delta y} = k = \text{const} \longrightarrow v_{i,M1} = v_{i,M2} + k\Delta y = v_{i,M2} + C$$

C is determined according to total mass conservation

$$\sum_{i=2}^{L2} \rho_{i,M1} (v_{i,M2} + C) \Delta x_i = \text{FLOWIN} \longrightarrow$$

$$C = \frac{\text{FLOWIN} - \sum \rho_{i,M1} v_{i,M2} \Delta x_i}{\sum \rho_{i,M1} \Delta x_i}$$

$v_{i,M1} = v_{i,M2}^* + C$ is taking as boundary condition for next iteration.

**When fully developed at outlet, : $f=1, C=0$;
Otherwise there is some differences between the two treatments, but such difference is not important.**

2) Outlet with recirculation

$\tilde{v}_{i,M1}$ is the normal velocity determined by local mass conservation, then

$$\sum_{i=2}^{L2} \rho_{i,M1} (f \cdot \tilde{v}_{i,M1}) \Delta x = FLOWIN_i \rightarrow f = \frac{FLOWIN}{\sum_{i=2}^{L2} \rho_{i,M1} \tilde{v}_{i,M1} \Delta x_i}$$

$v_{i,M1} = f \cdot \tilde{v}_{i,M1}$ is taking as the boundary condition for next iteration.

2. Applications

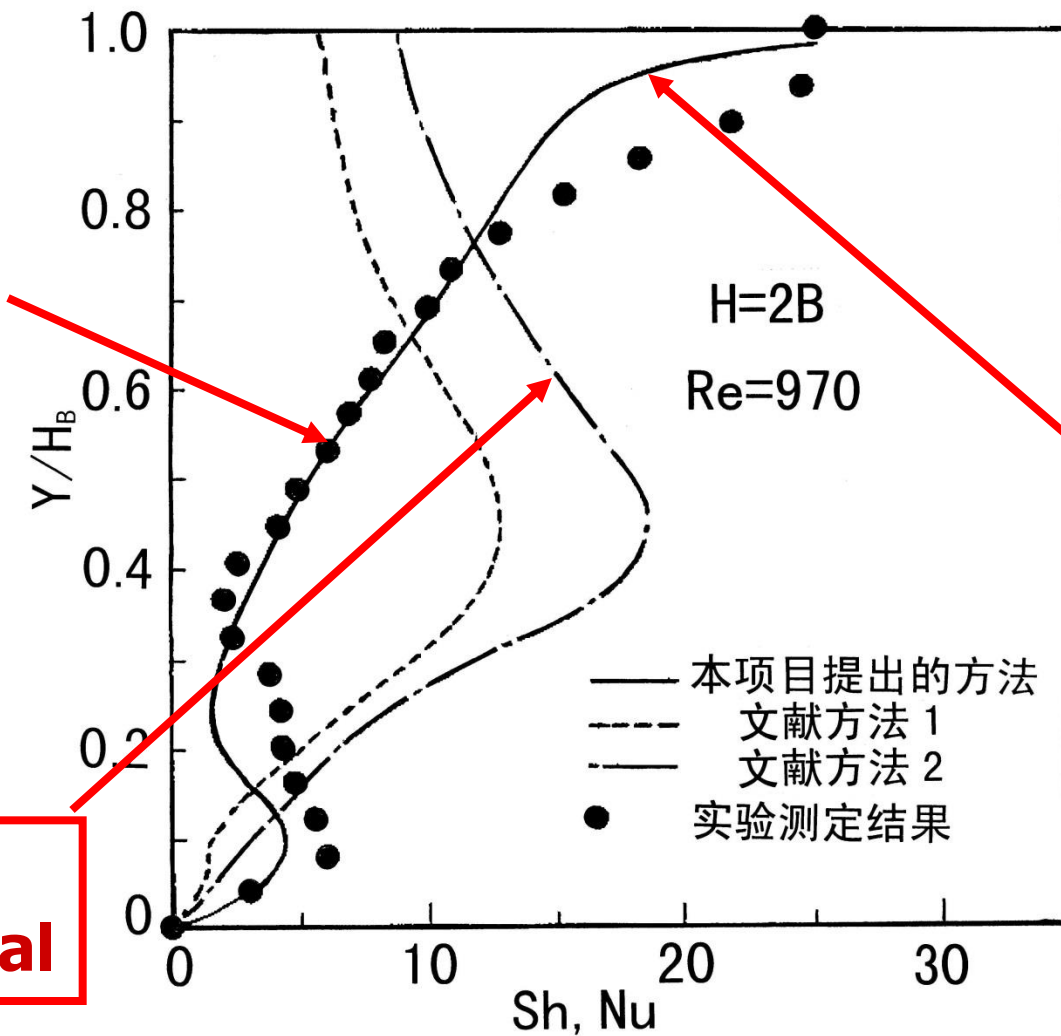
Li PW, Tao WQ. Effects of outflow boundary condition on convective heat transfer with strong recirculation flow. **Warme- Stoffubertrag**, 1994, 29 (8): 463-470

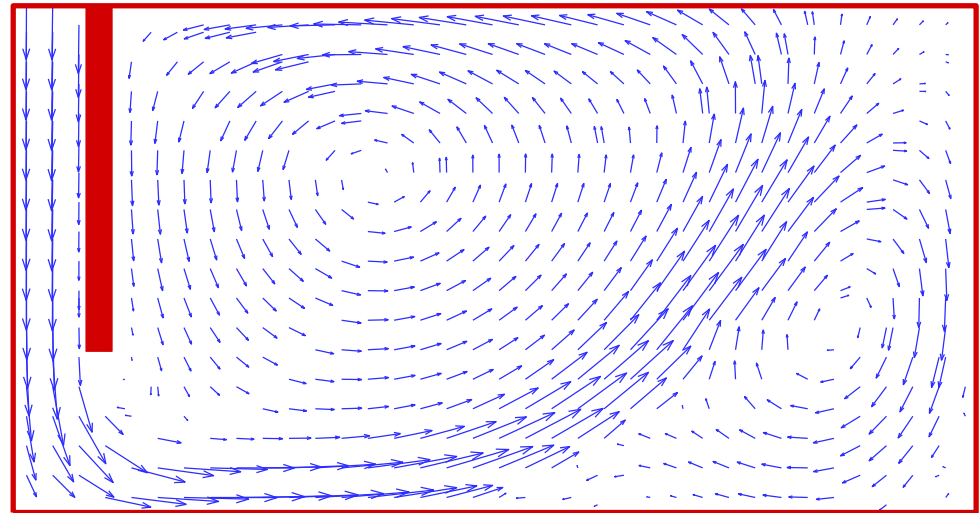
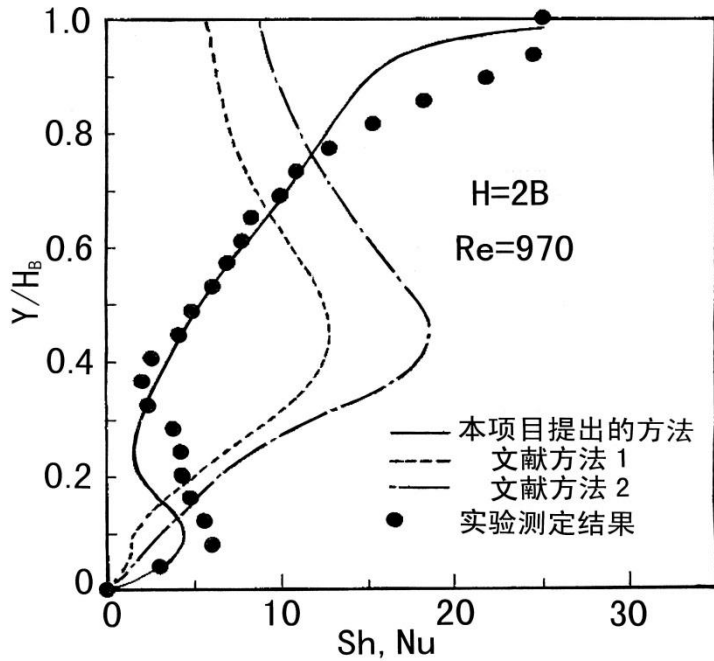
Comparison of predicted and measured local heat transfer coefficients

Tested

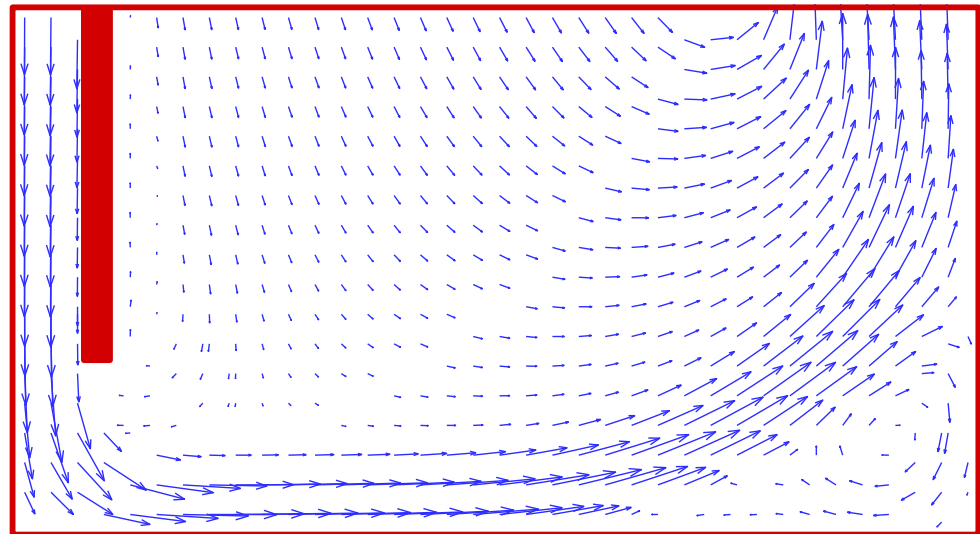
Other numerical

Mass conservation





(a) Mass conservation



(b) Reference

Effect of outflow boundary condition on flow pattern

6.9 Fluid Flow and Heat Transfer in a Closed system

6.9.1 Natural convection in enclosure

1. Boussinesq assumption
2. Governing eqs. of natural convection in enclosure
3. Effective pressure in natural convection in enclosure
4. Governing eqs. with Boussinesq assumption and effective pressure

6.9.2 Numerical treatments of island (孤島)

1. Method for setting zero velocity in island
2. Method for setting given temperature in island

6.9 Fluid Flow and Heat Transfer in a Closed system

6.9.1 Natural convection in enclosure

1. Boussinesq assumption

- 1) Viscous dissipation(耗散) is neglected;
- 2) Thermo-physical properties are constant except density;
- 3) Only the density in the gravitational term is considered varying with temperature as follows:

$$\rho = \rho_r [1 - \alpha(T - T_r)] \quad \alpha \text{ -expansion coefficient}$$

2. Governing equations of natural convection in enclosure

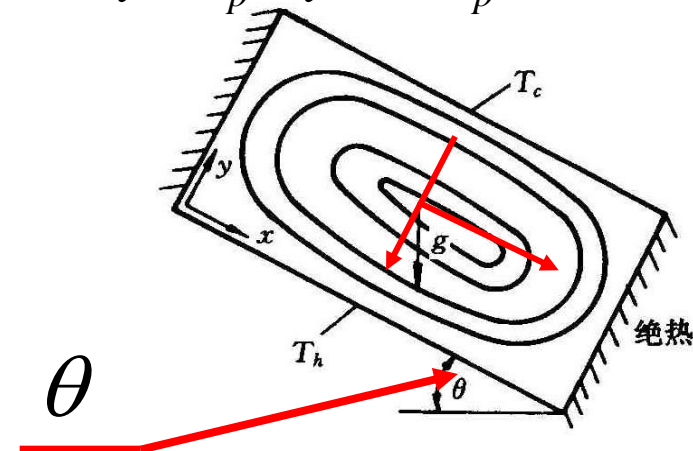
Governing equations for natural convection in a square cavity:

$$\frac{\partial(\rho u^2)}{\partial x} + \frac{\partial(\rho uv)}{\partial y} = -\frac{\partial p}{\partial x} + \frac{\partial}{\partial x} \left(\eta \frac{\partial u}{\partial x} \right) + \frac{\partial}{\partial y} \left(\eta \frac{\partial u}{\partial y} \right) + \rho g \sin \theta$$

$$\frac{\partial(\rho uv)}{\partial x} + \frac{\partial(\rho v^2)}{\partial y} = -\frac{\partial p}{\partial y} + \frac{\partial}{\partial x} \left(\eta \frac{\partial v}{\partial x} \right) + \frac{\partial}{\partial y} \left(\eta \frac{\partial v}{\partial y} \right) - \rho g \cos \theta$$

$$\frac{\partial(\rho uT)}{\partial x} + \frac{\partial(\rho vT)}{\partial y} = \frac{\partial}{\partial x} \left(\frac{\lambda}{c_p} \frac{\partial v}{\partial x} \right) + \frac{\partial}{\partial y} \left(\frac{\lambda}{c_p} \frac{\partial v}{\partial y} \right) + \frac{S_T}{c_p}$$

$$\frac{\partial(\rho u)}{\partial x} + \frac{\partial(\rho v)}{\partial y} = 0$$



3. Effective pressure in natural convection in enclosure

$$\frac{\partial(\rho u^2)}{\partial x} + \frac{\partial(\rho uv)}{\partial y} = -\frac{\partial p}{\partial x} + \frac{\partial}{\partial x} \left(\eta \frac{\partial u}{\partial x} \right) + \frac{\partial}{\partial y} \left(\eta \frac{\partial u}{\partial y} \right) + \rho g \sin \theta$$

$$-\frac{\partial p}{\partial x} + \rho g \sin \theta = -\frac{\partial p}{\partial x} + \rho_c g \sin \theta [1 - \alpha(T - T_c)]$$

$$= -\frac{\partial p}{\partial x} + g \rho_c \sin \theta - g \rho_c \alpha(T - T_c) \sin \theta$$

$$\frac{\partial(\rho uv)}{\partial x} + \frac{\partial(\rho v^2)}{\partial y} = -\frac{\partial p}{\partial y} + \frac{\partial}{\partial x} \left(\eta \frac{\partial v}{\partial x} \right) + \frac{\partial}{\partial y} \left(\eta \frac{\partial v}{\partial y} \right) - \rho g \cos \theta$$

$$-\frac{\partial p}{\partial y} - \rho g \cos \theta = -\frac{\partial p}{\partial y} - \rho_c g \cos \theta [1 - \alpha(T - T_c)]$$

$$= -\frac{\partial p}{\partial y} - g \rho_c \cos \theta + g \rho_c \alpha(T - T_c) \cos \theta$$

From $-\frac{\partial p}{\partial x} + g \rho_c \sin \theta$, $-\frac{\partial p}{\partial y} - g \rho_c \cos \theta$ an effective pressure is introduced:

$$p_{eff} = p - (g \rho_c \sin \theta)x + (g \rho_c \cos \theta)y$$

Then

$$\frac{\partial p_{eff}}{\partial x} = \frac{\partial p}{\partial x} - g \rho_c \sin \theta \quad \frac{\partial p_{eff}}{\partial y} = \frac{\partial p}{\partial y} + g \rho_c \cos \theta$$

The results is the same as in the momentum eqs.

Order of magnitude estimation(数量级估计) for $g \rho y$

For air: set $y=1\text{m}$, $g=9.8\text{ms}^{-2}$, $\rho = 1.2\text{kg} \cdot \text{m}^{-3}$

Then pressure introduced by natural convection is:

$$9.8 \times 1.2 \frac{\text{kg}}{\text{s}^2 \text{m}} = 11.76 \left(\frac{\text{kgm}}{\text{s}^2} \right) \left(\frac{1}{\text{m}^2} \right)$$

$$= 11.76 \frac{\text{N}}{\text{m}^2} = 11.76 \text{Pa}$$

4. G.Eqs. with Boussinesq assumption and effective pressure

Re-write ρ_c as ρ

$$\frac{\partial(\rho u^2)}{\partial x} + \frac{\partial(\rho uv)}{\partial y} = -\frac{\partial p_{eff}}{\partial x} + \frac{\partial}{\partial x} \left(\eta \frac{\partial u}{\partial x} \right) + \frac{\partial}{\partial y} \left(\eta \frac{\partial u}{\partial y} \right) - \rho g \alpha (T - T_c) \sin \theta$$

$$\frac{\partial(\rho uv)}{\partial x} + \frac{\partial(\rho v^2)}{\partial y} = -\frac{\partial p_{eff}}{\partial y} + \frac{\partial}{\partial x} \left(\eta \frac{\partial v}{\partial x} \right) + \frac{\partial}{\partial y} \left(\eta \frac{\partial v}{\partial y} \right) + \rho g \alpha (T - T_c) \cos \theta$$

$$\frac{\partial(\rho u T)}{\partial x} + \frac{\partial(\rho v T)}{\partial y} = \frac{\partial}{\partial x} \left(\frac{\lambda}{c_p} \frac{\partial v}{\partial x} \right) + \frac{\partial}{\partial y} \left(\frac{\lambda}{c_p} \frac{\partial v}{\partial y} \right) + \frac{S_T}{c_p}$$

$$\frac{\partial u}{\partial x} + \frac{\partial v}{\partial y} = 0$$

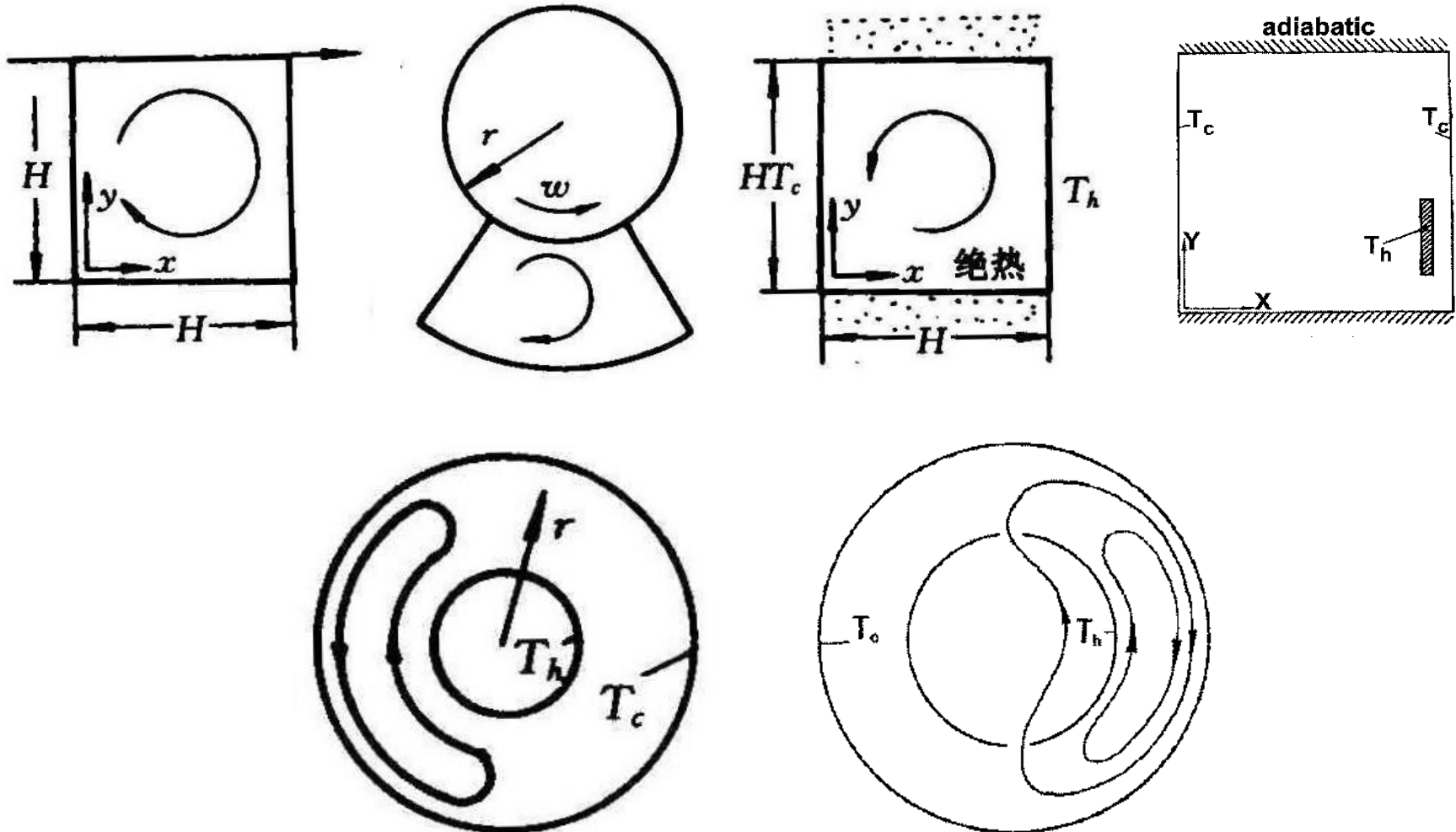
With correspondent boundary condition

5. Typical results of 2-D natural convection in enclosure

Table 6-8 2-D natural convection in enclosure of air

计算的量	$Ra = g\alpha(T_h - T_c)H^3 / \nu\lambda$			
	10^3	10^4	10^5	10^6
<u>Nu</u>	1.114	2.245	4.510	8.806
Nu_{\max}	1.581	3.539	7.637	17.442
$(Y/H)_{\max}$	0.099	0.143	0.085	0.036 8
Nu_{\min}	0.670	0.583	0.773	1.001
$(Y/H)_{\min}$	0.994	0.994	0.999	0.999
U_{\max}	0.153	0.193	0.132	0.077
$(Y/H)u_{\max}$	0.806	0.818	0.859	0.859
V_{\max}	0.155	0.234	0.258	0.262
$(X/H)v_{\max}$	0.181	0.119	0.066	0.039

6. Other examples of flow in enclosure



6.9.2 Numerical treatments for isolated island

Isolated island – solid region positioned within fluid region without connection with solid boundary.

An effective numerical method to numerically deal with the island is regarding the island as a special fluid region with very large viscosity.

1. Techniques guaranteeing zero velocity in island

(1) Setting zero values for u^0, v^0 in island at each iteration---Pay attention to the features of staggered grid system

- (2) Setting zero values for u^* , v^* in island---at each iteration the coefficients of the main diagonal of momentum equations are set to be infinite;
- (3) Setting near zero values for u' , v' in island---at each iteration set the values of d very small (10^{-25}) ;
- (4) Transferring near zero velocity in island to its boundary---at each iteration setting the diffusion coefficient very large (10^{25}) ; and adopting harmonic mean for interface interpolation.

2. Method for setting given temperature in island

(1) **Large coefficient method**—at each iteration resetting the coefficients in the **correspondent** discretized equations in island:

$$a_P \phi_P = \sum a_{nb} \phi_{nb} + b$$

$$a_P = A, b = A\phi_{given}, A = 10^{20} \sim 10^{30}$$

(2) **Large source term method**—at each iteration resetting source terms in island:

$$S_c = A\phi_{given}, S_P = -A, A = 10^{20} \sim 10^{30}$$

This method is effective only when $\alpha = 1$.

Remarks: In order to guarantee continuity of flux at the interface—the specific heat of solid region should take the value of fluid region.

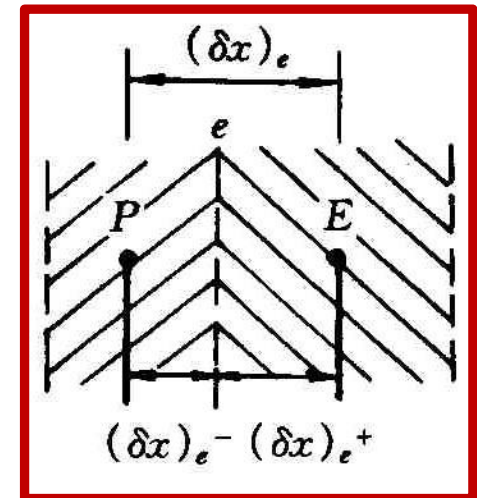
The harmonic mean for interface conductivity:

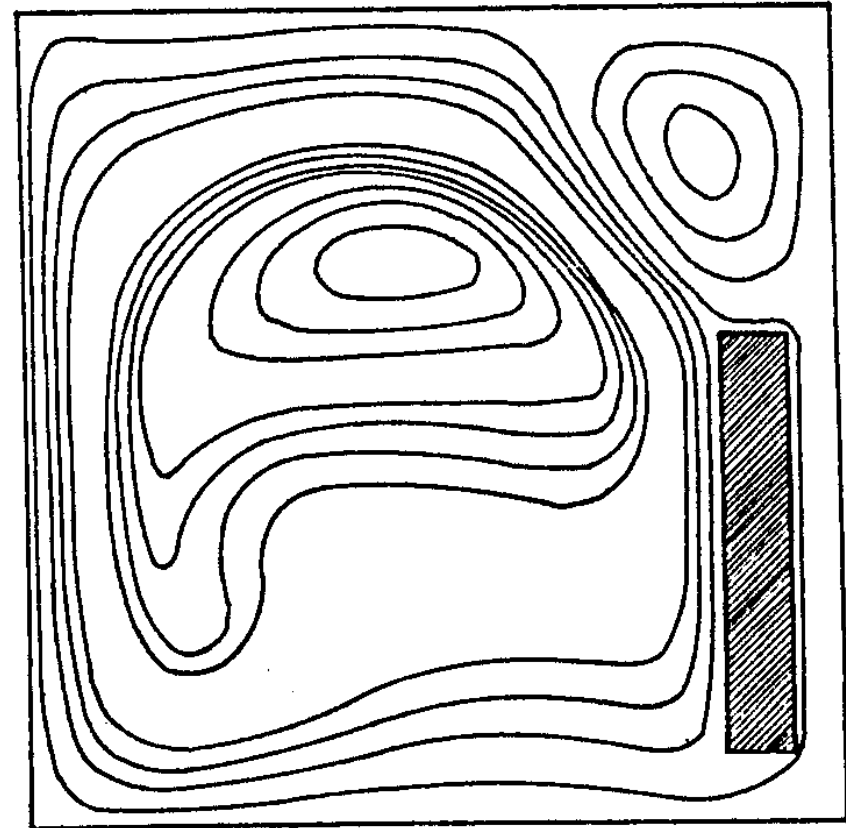
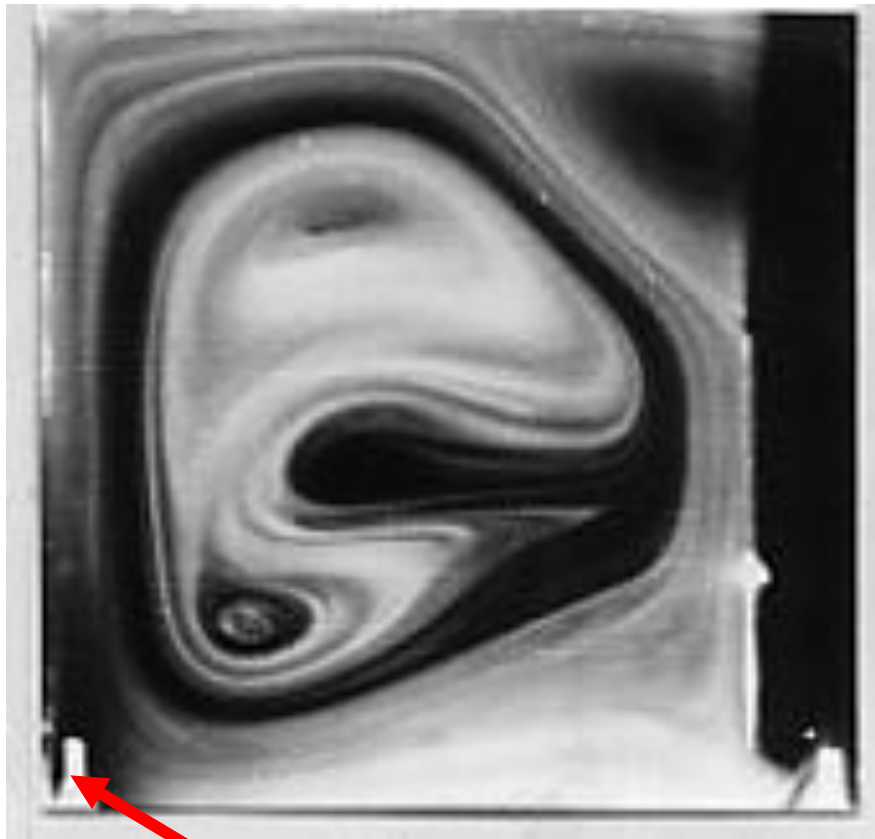
$$\frac{(\delta x)_e}{\lambda_e} = \frac{(\delta x)_{e^+}}{\lambda_E} + \frac{(\delta x)_{e^-}}{\lambda_P}$$

Now the normal diffusivity is used: $\Gamma = \lambda / c_p$

In order that harmonic mean is still valid for Γ , the specific heat must be the same at the two sides:

$$\frac{(\delta x)_e}{\lambda_e / (c_p)_f} = \frac{(\delta x)_{e^+}}{\lambda_E / (c_p)_f} + \frac{(\delta x)_{e^-}}{\lambda_P / (c_p)_f}$$





Gas inlet

Example of isolated island: comparison of numerical prediction and visualization (Hot island and cold enclosure wall)

Wang QW, Yang M, Tao WQ. Natural convection in a square enclosure with an internal isolated vertical plate. **Warme- Stoffubertrag**, 1994, 29 (3): 161-169

= Chapter 6 Primitive Variable Methods for Elliptic Flow and Heat Transfer

6.1 Source terms in momentum equations and two key issues in numerically solving momentum equation

6.2 Staggered grid system and discretization of momentum equation

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6.10 Introduction to collocated grid system

6.10 Introduction to Collocated Grid System

6.10.1 Success experience of staggered grid system
---adopting $1-\delta$ pressure difference and
possessing 2nd order accuracy (δ^2)

6.10.2 Discretization of momentum equation at
non-staggered grid system

1. Momentum conservation equation

2. Mass conservation equation

3. Key to collocated grid system---interface
velocity interpolation

6. Introduction to Collocated Grid System

6.10.1 Success experience of staggered grid system

1 – δ pressure difference can eliminate checkerboard pressure field

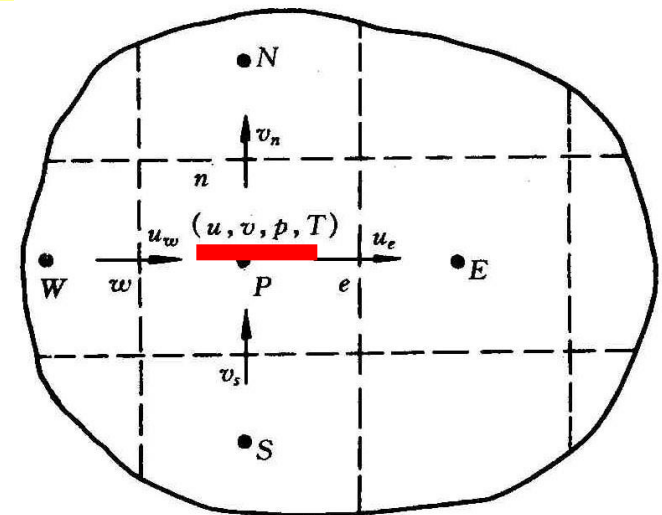
(δ^2) accuracy can guarantee diffusion feature of pressure equation.

6.10.2 Discretization at non-staggered grid system

1. Momentum equation

$$a_P u_P = \sum a_{nb} u_{nb} + b + A_P (p_w - p_e)$$

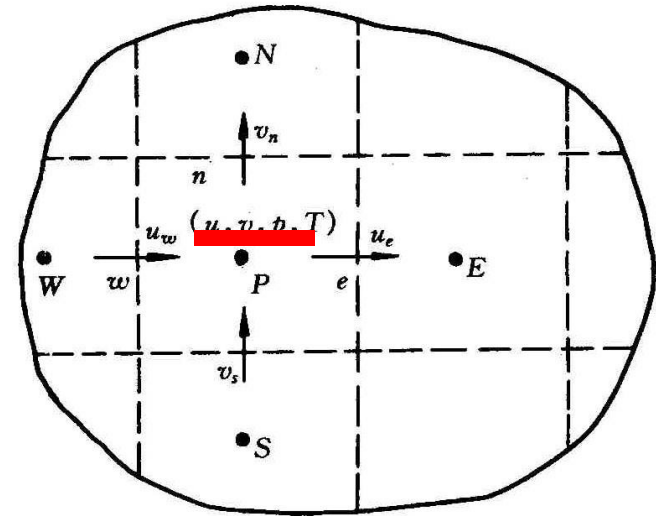
$$u_P = \frac{\sum a_{nb} u_{nb} + b}{a_P} + \frac{A_P}{a_P} (p_w - p_e)$$



$$u_P = u_P - \frac{A_P}{a_P} (p_e - p_w)$$

Similarly,

$$u_E = u_E - \left(\frac{A_P}{a_P} \right)_E (p_e - p_w)_E$$



Attention: pressure difference for velocity CV. E, the east-west interfaces are related to Node E

2. Mass conservation

For main grid CV. P, interface velocities are needed:

$$(\rho u A)_e - (\rho u A)_w + (\rho v A)_n - (\rho v A)_s = 0$$

3. Determination of interface velocity

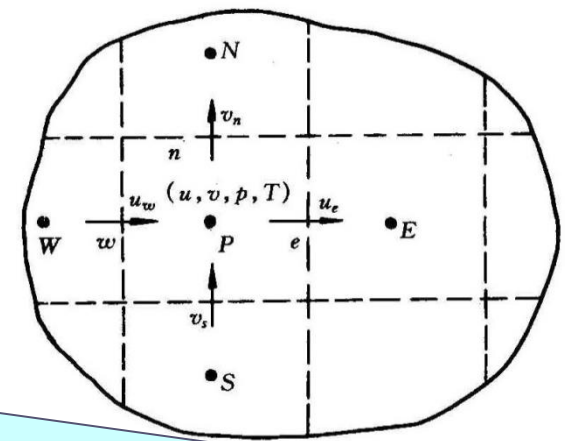
Determination of interface velocity provides an access (途径) for introducing $1 - \delta$ pressure difference

Mimicking (模仿) equations for u_P, u_E

$$u_E = u_E - \left(\frac{A_P}{a_P}\right)_E (p_e - p_w)_E$$

Interface velocity for e may be expressed as

$$u_e = u_e - \left(\frac{A_P}{a_P}\right)_e (\underline{p_E - p_P})$$



Pressure difference between two nodes over interface e

This is also the momentum equation for interface velocity.

This interpolation is called Momentum Interpolation Method-MIM, And is the key of success of collocated grid. Only for those non-

staggered grid systems which adopts MIM or the like are called collocated grid system (**同位网格**).

If the interpolation of interface velocity is simply taking the linear interpolation between u_P, u_E , then the chance for introducing 1-delta pressure difference is lost, and such non-staggered grid system can not work.

References

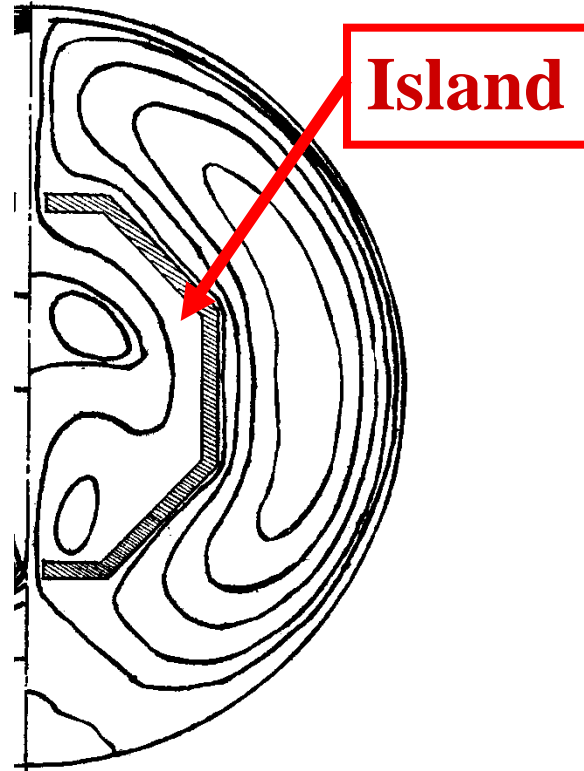
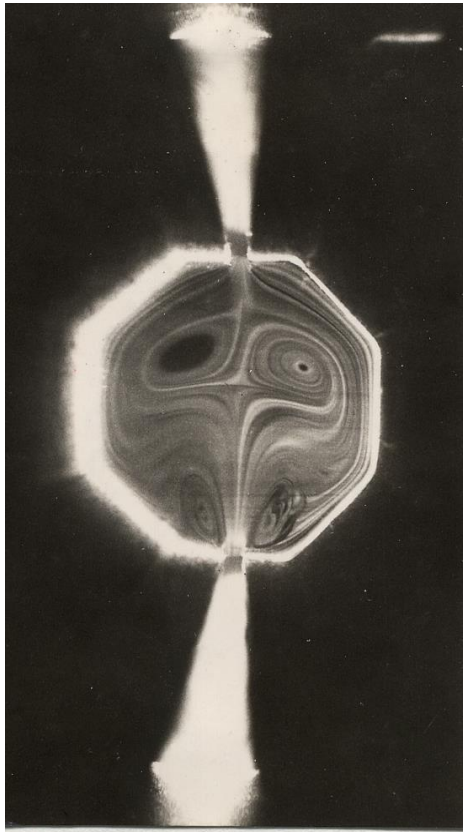
Peric M, Kessker R, Scheurere G. Comparison of finite numerical methods with staggered and collocated grids. **Computers & Fluids**. 1988, 16:389-403

Yu B, Kawaguchi Y, Tao WQ, Ozoe H. Checkerboard pressure predictions due to the underrelaxation factor and time step size for a nonstaggered grid with momentum interpolation method. **Numerical Heat Transfer, Part B**, 2002, 41 (1): 85-94

Yu B, Tao WQ, Wei JJ, et al. Discussion on momentum interpolation method for collocated grids of incompressible flow. **Numerical Heat Transfer, Part B**, 2002, 42 (2): 141-166

Qu ZG, Tao WQ, He YL. Implementation of CLEAR algorithm on collocated grid system and application examples. **Numerical Heat Transfer, B**, 2005 (1):65-96.

Appendix of Section 6-9



Visualization results

Numerical results

Visualization results

Comparison of predicted and visualized natural convection in large electric current bus bar

Home work

6-1 6-4 6-5 6-6 6-9

Due in Nov.16

Problem # 6-1⁴

- As mentioned in section 6.1, the problem of segregated algorithm for fluid flow, there is no independent governing equation for pressure. In order to deal with the problem of the coupling between the pressure and velocity, SIMPLE and a series of algorithms are introduced. But, on the other hand, the pressure Poisson equation can be derived from the momentum equation and continuity equation, for example, as shown below is the equation of two-dimensional rectangular coordinates for incompressible fluid:⁴

•⁴

$$\frac{\partial^2 p}{\partial x^2} + \frac{\partial^2 p}{\partial y^2} = 2 \left[\left(\frac{\partial u}{\partial x} \right) \left(\frac{\partial v}{\partial y} \right) - \left(\frac{\partial v}{\partial x} \right) \left(\frac{\partial u}{\partial y} \right) \right]^{4}$$

- Somebody thinks that we can get the pressure equation and momentum equation simultaneous to solve the flow, namely, solving the u equation, the v equation and pressure equation in turn (u , v are known, and can be the source term of the pressure equation). This is the one iteration of the separation method, thus, without using SIMPLE algorithm. Try to derive the pressure Poisson equation and comment on the view.⁴

Problem # 6-4

- Considering the two-dimensional flow in figure 6-11, whereas $u_w = 50$, $v_s = 20$, $p_N = 0$, $p_E = 10$ are known, assuming the flow is in steady state and the density is constant. The discrete equation for u_e , v_n is:
 - $u_e = p_P - p_E$; $v_n = 0.7(p_P - p_N)$
- Using the SIMPLE algorithm to get the value of u_e , v_n and p_P .

Problem # 6-5

- A piping system as shown in figure 6-33 is applied to pump fluid from node 1 to node 2, 3, 4, 5, 6, 7. The pressure of node 1,2,4,5 is given in the parentheses. The flow rate between two nodes can be calculated by using formula $Q = C(\Delta p)$, where Δp is the pressure difference between the two nodes and C is the hydraulic conductivity. For simplicity, the conductivity of two adjacent nodes is shown by using the letter on the

midpoint of the two nodes as a subscript. Such as C_D could be expressed as the hydraulic conductivity between the nodes 3, 6. Where $C_A = 0.4$, $C_B = 0.2$, $C_C = 0.1$, $C_D = 0.2$, $C_E = 0.1$, $C_F = 0.2$ are known. The flow rate between the 6, 7 is $Q_F = 20$. All quantities are in same unit system. Try to use similar SIMPLE algorithm to determine the value of p_3 , p_6 , Q_A , Q_B , Q_C , Q_D , Q_D and p_7 (Hint : firstly assume p_3^* , p_6^* to get the flow rate and then calculate the pressure correction value using the mass conservation between node 3, 6)↵

Problem # 6-6↵

- Consider one-dimensional flow within the porous medium. Governing equations are

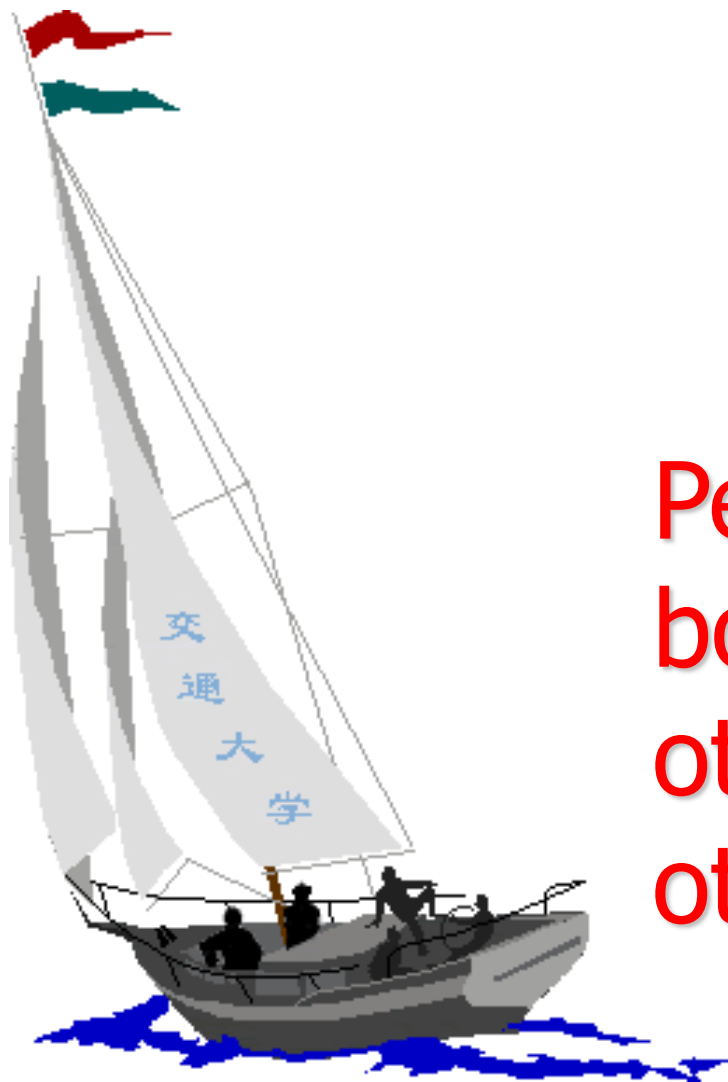
$$c|u|u + \frac{dp}{dx} = 0 \quad \text{and} \quad \frac{d(uA)}{dx} = 0, \quad \text{where } c \text{ is constant, } A \text{ is the effective area. For a}$$

discrete system shown in figure 6- 12, $\Delta x = 1$ (uniform grid), $C_B = 0.4$,

$C_C = 0.2$ $A_B = 2$, $A_C = 3$, $p_1 = 140$, $p_3 = 30$ are known. All units are coordinated. Use

SIMPLE algorithm to get the value of p_2 , u_B , u_C ↵

Problem# 6-9 As shown in Fig. 6-35 laminar flow and heat transfer in a straight duct with square cross section are fully developed. The duct top and bottom surfaces are adiabatic, while its right and left walls are at T_h and T_c , respectively. Adopting Boussinesq assumption and effective pressure concept, and considering the effect of buoyancy force, write the governing equations of the three velocity components in x, y and z direction denoted by u, v, w respectively.



同舟共济 渡彼岸!

People in the same
boat help each
other to cross to the
other bank, where....